

工學碩士 學位論文

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Vehicle Control and Performance Analysis of 4WS  
Passenger Cars using Robust Control Techniques

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2000年 2月

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## **Abstract**

In this dissertation, a lateral control design is presented for automatic steering of active four-wheel steering (4WS) vehicles for highway driving. The linearized two degree-of-freedom (2 DOF) equations for the lateral dynamics are derived using the Newton's equations. A robust controller using  $\mu$ -analysis and synthesis is designed for a linear model of a passenger cars moving a given path. The performance of the robust controller is then evaluated using simulation studies.

It is shown that the presented control method possesses the inherent advantages that are robust to complex uncertainty for typical driving maneuvers. Finally, the active 4WS vehicle achieves good performance for a wide range of uncertainty in the highway operating conditions.

$\beta$	:			
$\dot{r}$	:		$(\dot{r} = \frac{d}{dt} \phi)$	
$C_{\alpha f}$	:			
$C_{\alpha r}$	:			
$M$	:			
$V$	:	CG		
$I_{zz}$	:	z		
$l_f (l_r)$	:	( )	CG	
$l_w$	:			CG
$f_w$	:			
$\delta_f (\delta_r)$	:	( )		
$\alpha_f (\alpha_r)$	:	( )		
$L = l_f + l_r$	:			
$a_y$	:		가	
$F_f (F_r)$	:	( )		
$A^*$	:	A		
$\overline{\sigma}(A)$	:	A		
$\underline{\sigma}(A)$	:	A		
$trace(A)$	:	A		
$\rho(A)$	:	A		(Spectral Radius)

$\Delta$  :

$$G(s) := \left[ \begin{array}{c|c} A & B \\ \hline C & D \end{array} \right] := C (sI - A)^{-1} B + D$$

# 1

## 1.1

Two Wheel Steering(2WS), Four  
Wheel Steering(4WS) . Ackermann

가

4WS Model

. [11] [16] 4WS

(  $r$  ) 가 (accelerometer) 가

(  $a_y$  ), (  $V$  )

가

. [7], [8], [1

6] [18]

(dynamic

compensation)

(yawing)

가

(phase lag control)

(phase lead control) ,

가

가



. [8] [10], [17] [19]

## 1.2

$H_\infty$   $\mu$ - Analysis and Synthesis

(model uncertainty)

.  
Bicycle Model (single  
track model)  
(disturbance),

$H_\infty$   $\mu$

가 .

가

가

$$H_{\infty} \quad \mu$$

### 1.3

, 가  
 (lateral acceleration) . ,  
 (disturbance) . ,  
 (vehicle  
 sideslip angle) . .  
 . , 가  
 (robust stability)  
 (robust performance) 가 .  
 (lateral stability) (ride comfort)

## 2

### 2.1 Bicycle Model (Single-Track Model)

, Bicycle

Fig.1

(yaw motion)

(side slip motion) 2

Fig.1 Single Track Model

Fig.2

가 Fig.2

. ([7] [10], [17] [18])

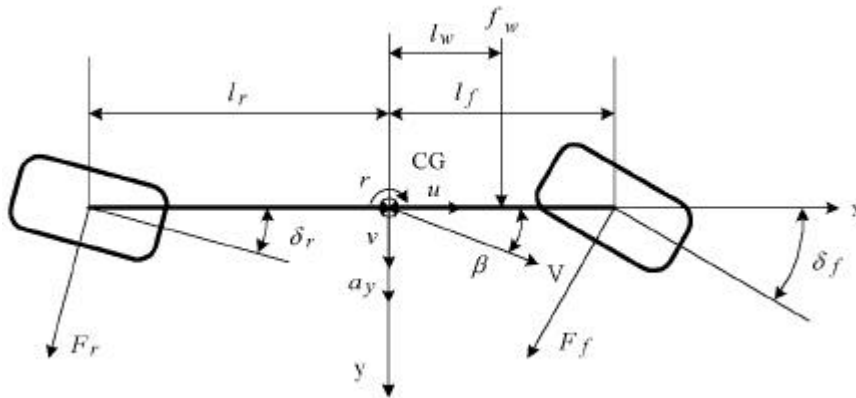


Fig.1 Single-track vehicle model

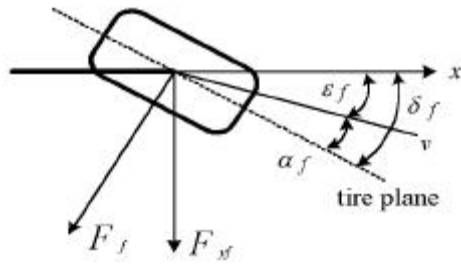


Fig.2 Tire slip angle diagram  
(front tire)

## 2.2

가

가 ,

가

(lateral acceleration) 가 가 .

Tire Sideslip Angle ( $\alpha$ ) ( $F_y$ )

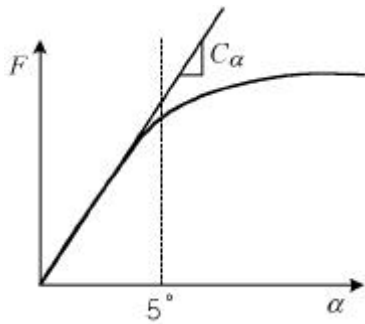


Fig.3 Cornering force characteristics

(cornering force)

Side Slip Angle

가 . Fig.3

$$\alpha \leq 5^\circ$$

$\alpha$ 가 가  $F$  가

. ([4] pp. 198 200, [6] pp. 29 33)

$$(1)$$

$$F = C_{\alpha} \alpha \quad (1)$$

(1)  $C_{\alpha}$  가 Cornering

Stiffness . (2)

$$F_{yf} = C_{\alpha f} \alpha_f$$

$$F_{yr} = C_{\alpha r} \alpha_r \quad (2)$$

가 가  $6 \text{ m/s}^2$  가

가  $5^\circ$  가

가

### 2.3 SAE

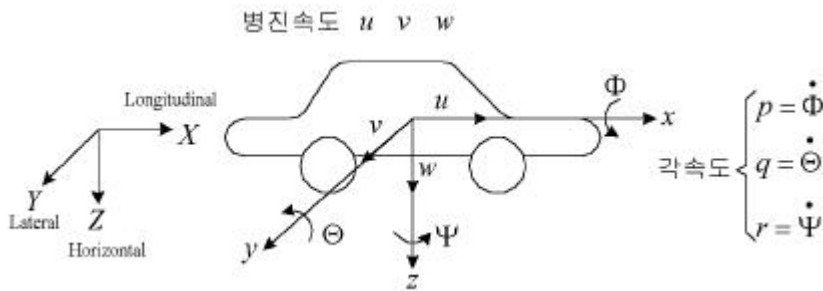


Fig.4 Vehicle axis system (6DOF )

Fig.4 3 6

$x, y, z$

$u, v, w$  가

roll(  $\Phi$ ), pitch(  $\Theta$ ), yaw(  $\Psi$ )가 .

$p, q, r$  .

## 2.4 2

(yaw motion)

(sideslip motion) 2

(side slip angle) (3) (5)

. Fig.2 (  $\alpha_f$  ) (3) .

$$\alpha_f = \delta_f - \epsilon_f \quad (3)$$

$$\epsilon_f = \tan^{-1} \left( \frac{v + l_f r}{u} \right)$$

,  $\epsilon_f$  가 가

$$\epsilon_f = \tan^{-1} \left( \frac{v + l_f r}{u} \right) \simeq \left( \frac{v + l_f r}{u} \right) \text{ 가 } .$$

$$\alpha_f = \delta_f - \epsilon_f = \delta_f - \left( \frac{v + l_f r}{u} \right) \quad (4)$$

가 (4) (5)가

$$\alpha_r = \delta_r - \varepsilon_r = \delta_r - \left( \frac{l_f r - v}{u} \right) \quad (5)$$

Fig.1 Single Track Model Newton 2

y

$$\downarrow + \sum \text{Force} = M \cdot a_y \quad (6)$$

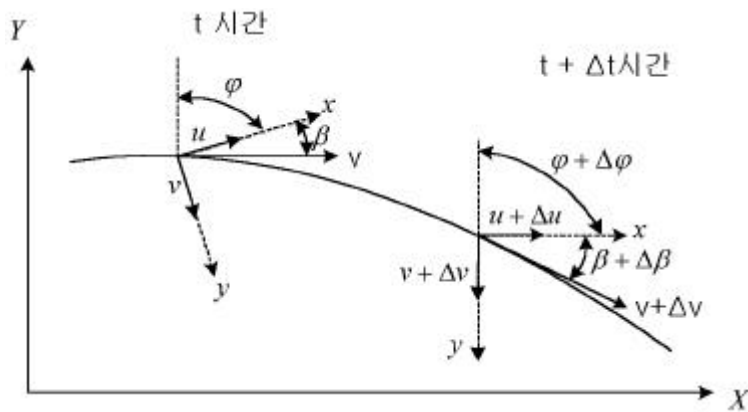


Fig.5 Transient planar motions of a vehicle using axes fixed to chassis

가  $a_y$  Fig.5 (7)

$$\begin{aligned} a_y &= \lim_{\Delta t \rightarrow 0} \frac{v' - v}{\Delta t} = \lim_{\Delta t \rightarrow 0} \frac{-\Delta v + u \Delta \phi}{\Delta t} \\ &= \lim_{\Delta t \rightarrow 0} \frac{-\Delta v}{\Delta t} + u \lim_{\Delta t \rightarrow 0} \frac{-\Delta \phi}{\Delta t} \\ &= \frac{dv}{dt} + u r = \dot{v} + u r \end{aligned} \quad (7)$$

가 (lateral acceleration) (8)

$$a_y = \dot{v} + ur \quad (8)$$

$$(6) \quad (9)$$

$$M(\dot{v} + ur) = F_{yf} + F_{yr} + f_w \quad (9)$$

$$(2) \quad (10)$$

$$M\dot{v} + Mur = \left( \frac{-C_{\alpha f} - C_{\alpha r}}{u} \right)v - \left( \frac{C_{\alpha f}l_f - C_{\alpha r}l_r}{u} \right)r + C_{\alpha f}\delta_f + C_{\alpha r}\delta_r + f_w \quad (10)$$

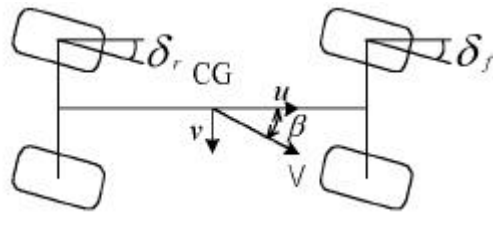


Fig.6 Planar model of 4WS car with steer angle inputs

, Fig.6  $\beta$  가

$$v = \beta V$$

$$\dot{v} = \dot{\beta}V$$

$$u \approx V \quad (11)$$

(10) Vehicle Sideslip



$$MV \dot{\beta} = - (C_{cf} + C_{cr})\beta + \left( -MV - \frac{C_{cf}l_f - C_{cr}l_r}{V} \right) r + C_{cf}\delta_f + C_{cr}\delta_r + f_w \quad (12)$$

(12) Side Slip

$$\dot{\beta} = - \left( \frac{C_{cf} + C_{cr}}{MV} \right) \beta + \left( -1 - \frac{C_{cf}l_f - C_{cr}l_r}{MV^2} \right) r + \frac{C_{cf}}{MV} \delta_f + \frac{C_{cr}}{MV} \delta_r + \frac{1}{MV} f_w \quad (13)$$

Yaw

z

$$\dot{\psi} + \sum \text{Moment} = I_{zz} \dot{r} \quad (14)$$

Fig.1

$$I_{zz} \dot{r} = l_f F_{yf} - l_r F_{yr} + l_w f_w \quad (15)$$

(2) (4),(5) (15)

$$= l_f \left( C_{cf} \left( \delta_f - \frac{v + l_f r}{u} \right) \right) - l_r \left( C_{cr} \left( \delta_r - \frac{l_r r - v}{u} \right) \right) + l_w f_w \quad (16)$$

(13) 가 Fig.5 (11) (16) Yaw

$$I_{zz} \dot{r} = - (l_f C_{cf} \delta_f - l_r C_{cr} l_r) \beta - \left( \frac{C_{cf} l_f^2 + C_{cr} l_r^2}{V} \right) r + C_{cf} l_f \delta_f - C_{cr} l_r \delta_r + l_w f_w \quad (17)$$

Yaw

(18)

$$\dot{r} = - \left( \frac{l_f C_{cf} \delta_f - l_r C_{cr} l_r}{I_{zz}} \right) \beta - \left( \frac{C_{cf} l_f^2 + C_{cr} l_r^2}{I_{zz} V} \right) r$$

$$+ \frac{C_{\alpha f} l_f}{I_{zz}} \delta_f - \frac{C_{\alpha r} l_r}{I_{zz}} \delta_r + \frac{l_w}{I_{zz}} f_w \quad (18)$$

([4]pp. 195-235, [5]pp. 213-241)

## 2.5.2

$$(13) \quad (18)$$

2, (side

slip) (yaw rate)

가

$$x = \begin{bmatrix} \beta \\ r \end{bmatrix}$$

가

( $f_w$ ),

( $\delta_f$ ), ( $\delta_r$ )

$$u = [f_w \quad \delta_f \quad \delta_r]^T$$

(19)

$$\dot{x} = Ax + Bu \quad (19)$$

(19) A Matrix B Matrix (20), (21)

$$A = \begin{bmatrix} -\frac{(C_f + C_r)}{MV} & -1 - \frac{(C_f l_f - C_r l_r)}{MV^2} \\ -\frac{(C_f l_f - C_r l_r)}{I_{zz}} & -\frac{(C_f l_f^2 + C_r l_r^2)}{I_{zz} V} \end{bmatrix} \quad (20)$$

$$B = \begin{bmatrix} \frac{1}{MV} & \frac{C_f}{MV} & \frac{C_r}{MV} \\ \frac{l_w}{I_{zz}} & \frac{C_f l_f}{I_{zz}} & \frac{C_r l_r}{I_{zz}} \end{bmatrix} \quad (21)$$

가  $(a_y)$  Side Slip Angle( $\beta$ )

$$y = \begin{bmatrix} a_y \\ \beta \end{bmatrix}$$

$$y = Cx + Du \quad (22)$$

(22) C Matrix D Matrix (23), (24)

$$C = \begin{bmatrix} \frac{(C_f + C_r)}{M} + \frac{l_f(l_f C_r - l_r C_f)}{I_{zz}} & 1 \\ \frac{(l_f C_f - l_r C_r)}{MV} + \frac{l_f(l_f^2 C_f + l_r C_r)}{I_{zz} V} & 0 \end{bmatrix} \quad (23)$$

$$D = \begin{bmatrix} \frac{(l_r + l_w)}{M l_r} & -\frac{C_f}{MV} + \frac{l_f^2 C_f}{I_{zz}} & \frac{C_r}{MV} + \frac{l_r^2 C_r}{I_{zz}} \\ 0 & 0 & 0 \end{bmatrix} \quad (24)$$

### 3

#### 3.1 (Norm)

Norm 가

Signal Norm, Matrix Norm, System Norm .

Signal Norm

1-Norm :  $e(t)$  (25-1)

$$\| e \|_1 := \int_{-\infty}^{\infty} |e(t)| dt \quad (25-1)$$

$e(t)$  2-Norm (25-2)

$$\| e \|_2 := \sqrt{\left( \int_{-\infty}^{\infty} e(t)^2 dt \right)} \quad (25-2)$$

$e(t)$  2-Norm (25-3)

$$\| e \|_2 = \sqrt{\left[ \frac{1}{2\pi} \int_{-\infty}^{\infty} |e(jw)|^2 dw \right]} \quad (25-3)$$

$e(t)$   $\infty$ -Norm :  $e(t)$  가

$$\| e \|_{\infty} = \sup_t | e(t) | \quad (25-4)$$

*sup* Supremum Least Upper Bound

$e(t)$  가 Vector Signal Vector Norm ,

$$e(t) = \begin{bmatrix} e_1(t) \\ e_2(t) \\ \vdots \\ e_n(t) \end{bmatrix} \in \mathbf{R}^n$$

Vector Signal 2-Norm (25-5)

$$\|e\|_2 := \sqrt{\left(\int_{-\infty}^{\infty} |e(t)|^2 dt\right)} = \sqrt{\left(\int_{-\infty}^{\infty} e^T(t)e(t) dt\right)} \quad (25-5)$$

Matrix Norm

Matrix Norm Induced Norm, (26-1)

$$\|A\|_p := \sup_{x \neq 0} \frac{\|Ax\|_p}{\|x\|_p} \quad (26-1)$$

2-Norm (p=2) (26-2)

$$\|A\|_2 = \sqrt{\lambda_{\max}(A^*A)} \quad (26-2)$$

, 2-Norm A (maximum singular value)

$$\|A\|_2 = \bar{\sigma}(A)$$

Maximum Singular Value 3.2

$\infty$ -Norm (p=∞) (26-3)

$$\|A\|_{\infty} = \max_i \sum_{j=1}^n |a_{ij}| \quad (26-3)$$

System Norm

(27)

$$\begin{bmatrix} \dot{x} \\ y \end{bmatrix} = \begin{bmatrix} A & B \\ C & D \end{bmatrix} \begin{bmatrix} x \\ u \end{bmatrix} \quad (27)$$

$$y(s) \quad u(s)$$

$$G(s) \quad (27-1)$$

$$y(s) = G(s)u(s) \quad (27-1)$$

$$G(s) \quad (27-2)$$

$$G(s) := C(sI - A)^{-1}B + D \quad (27-2)$$

$H_2$  Norm     $H_\infty$  Norm    (28-1), (28-2)

$H_2$  Norm

$$\|G\|_2 := \sqrt{\left[ \frac{1}{2\pi} \int_{-\infty}^{\infty} \text{trace } G^*(j\omega)G(j\omega) d\omega \right]} \quad (28-1)$$

$$\text{trace}(A) \quad \text{trace}(A) := \sum_{i=1}^n a_{ii}$$

$H_\infty$  Norm

$$\|G\|_\infty := \max_{\omega \in \mathbf{R}} \bar{\sigma}[G(j\omega)] \quad (28-2)$$

$H_\infty$  Norm     $G(s)$     가

### 3.2 (Singular Value Decomposition)

Singular Value Decomposition    A

(largest singular value) (smallest singular value)

$$A \in \mathbb{C}^{m \times n} \quad U \quad V \quad (29-1), (29-2)$$

$$U = [u_1, u_2, \dots, u_m] \in \mathbb{C}^{m \times m} \quad (29-1)$$

$$V = [v_1, v_2, \dots, v_n] \in \mathbb{C}^{n \times n} \quad (29-2)$$

$U$   $m \times m$  Unitary,  $V$   $n \times n$  Unitary.

$$A \quad (30)$$

$$A = U \Sigma V^* \quad (30)$$

$$(30) \quad \Sigma_1 \quad A$$

(singular value)

$$(30) \quad \Sigma = \begin{bmatrix} \Sigma_1 & 0 \\ 0 & 0 \end{bmatrix}$$

$$\Sigma_1 = \begin{bmatrix} \sigma_1 & 0 & \dots & 0 \\ 0 & \sigma_2 & \dots & 0 \\ \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & \dots & \sigma_p \end{bmatrix}$$

$$\sigma_1 \geq \sigma_2 \geq \dots \geq \sigma_p \geq 0, \quad p = \min\{m, n\} \quad (30-1)$$

$$(30-1) \quad A \quad \sigma_1$$

$$\sigma_1 = \sigma_{\max}(A) = \bar{\sigma}(A)$$

$$\bar{\sigma}(A) = \sqrt{\lambda_{\max}(A^* A)}$$

$\lambda_{\max}$

,  $A$   $\sigma_p$   $\sigma_p = \sigma_{\min}(A) = \underline{\sigma}(A)$

$\underline{\sigma}(A) = \sqrt{\lambda_{\min}(A^*A)}$ ,  $\lambda_{\min}$

### 3.3 (Plant Uncertainty)

가 , (nominal)

가

1) (Parametric Uncertainty)

가

가



2) (Unmodeled Dynamics)

,  
 ,  
 가 .

3)

(Additive Uncertainty)

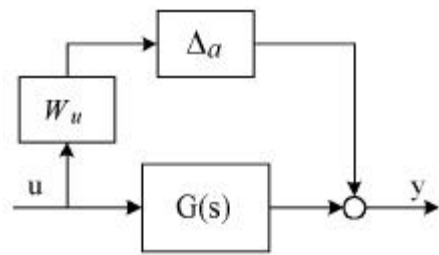


Fig.7 Plant uncertainty with additive perturbation

Fig.7

$$y = (G + \Delta_a W_u)u \quad \|\Delta_a\|_\infty \leq 1$$

$$\tilde{G}(s) = G(s) + \Delta_m W_m$$

(Multiplicative Uncertainty)

Fig.8 Fig.9

가

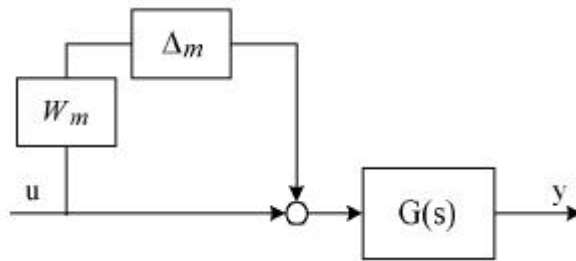


Fig.8 Plant uncertainty with input multiplicative perturbation

Fig.8

$$y = G (1 + W_m \Delta_m) u, \quad \|\Delta_m\|_\infty \leq 1$$

$$\tilde{G} = G + G W_m \Delta_m$$

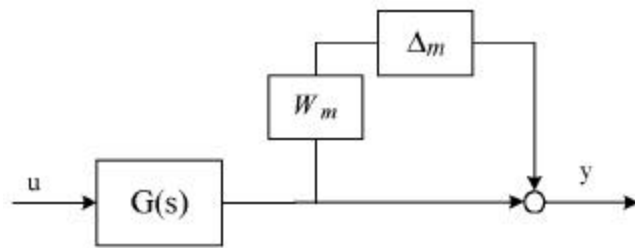


Fig.9 Plant uncertainty with output multiplicative perturbation

Fig.9

$$y = (1 + W_m \Delta_m) Gu, \quad \|\Delta_m\|_\infty \leq 1$$

$$\tilde{G} = G + W_m \Delta_m G$$

### 3.4

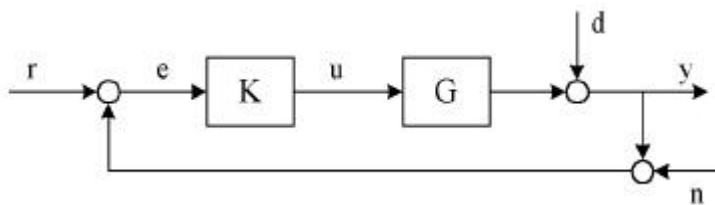


Fig.10 Classical closed-loop feedback system

Fig.10

.  $P(s)$

. Fig.11

$\Delta$

,

$K$

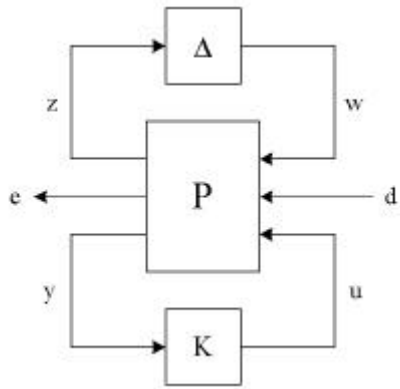


Fig.11 General Framework for robust performance

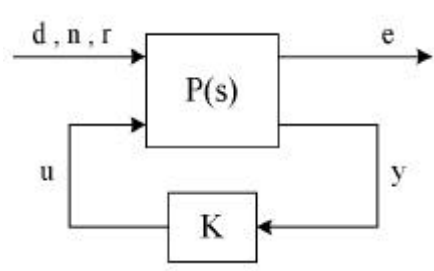


Fig.12 Control synthesis framework

$\mu$

Fig.10

Fig.12

. Fig.12

Fig.13

$P(s)$

Open Loop Interconnection

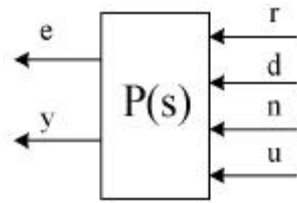


Fig.13 Open-loop interconnection matrix

### 3.5 Open Loop Interconnection

Fig.11

Fig.12

System Matrix  $P(s)$

$K$

가 .

$$\begin{bmatrix} e \\ y \end{bmatrix} = P \begin{bmatrix} d \\ u \end{bmatrix} = \begin{bmatrix} P_{ed} & P_{eu} \\ P_{yd} & P_{yu} \end{bmatrix} \begin{bmatrix} d \\ u \end{bmatrix} \quad (31)$$

$$(31) \quad d \quad e \quad (T_{ed}) \quad (32)$$

$$\frac{e}{d} = P_{ed} + P_{eu}K(I - P_{yu}K)^{-1}P_{yd} = T_{ed} \quad (32)$$

### 3.6 LFT (Linear Fractional Transformation)

LFT Open Loop Interconnection

System Matrix

$K$

Lower LFT System Matrix

Upper LFT      가      가      .

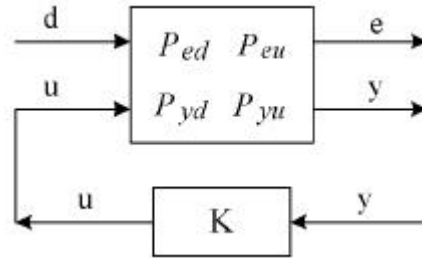


Fig.14 Generalized plant with lower LFT

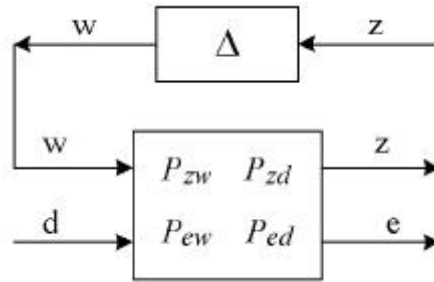


Fig.15 Generalized plant with upper LFT

1) Lower LFT

Fig. 14

$$e = P_{ed}d + P_{eu}u$$

$$y = P_{yd}d + P_{yu}u$$

$$u = Ky$$

가      .

$u \quad y \quad d \quad e$

$$e = [P_{ed} + P_{eu}K(I - P_{yu}K)^{-1}P_{yu}]d \quad (33-1)$$

(33-1) Lower LFT  $F_L(P, K)$  .

## 2) Upper LFT

Fig. 15

$$z = P_{zw}w + P_{zd}d$$

$$e = P_{ew}w + P_{ed}d$$

$$w = \Delta z$$

가

$w \quad z \quad d$

$e$

$$e = [P_{ed} + P_{ew}\Delta(I - P_{zw}\Delta)^{-1}P_{zd}]d \quad (33-2)$$

(33-2) Upper LFT  $F_U(\Delta, P)$  .

## 3.7 가 (Weighting Functions)

1)

가

가 가

(magnitude) ,

가 .

.

2) 가

$$, W = \frac{1}{s+1} \quad (34-1)$$

(32-1) 가 .

Minimum Phase

$$W = \frac{s+1}{(s+2)(s+3)} \quad (34-2)$$

(32-2)  $s$

.

가 가 .

### 3.8 $H_\infty$ Problem

1)  $H_\infty$  Optimal Problem

$$\min_{K(s)} \|T_{ed}\|_\infty = \gamma_{opt} \quad (35)$$

(35)  $d$   $e$   $\infty$

- Norm  $K$  .  $T_{ed}$



$d$   $e$   $\cdot$   $H_\infty$  Optimal  
 Problem  $\gamma_{opt}$   $\cdot$   
 Control Energy가  
 $\cdot$   $H_\infty$  Optimal Controller  
 (uncertainty) 가  
 $\cdot$   $\mu$   
 $H_\infty$  Optimal  
 Controller  $\cdot$

2)  $H_\infty$  Suboptimal Problem

$$\text{Find } \|T_{ed}(s)\|_\infty \leq \gamma \quad (35-1)$$

$H_\infty$  Optimal Problem

$$(35-1) \quad \gamma$$

가  $K$

$$\begin{bmatrix} \text{Tracking Errors} \\ \text{Control Inputs} \end{bmatrix} = T \begin{bmatrix} \text{Reference Inputs} \\ \text{Disturbances} \\ \text{Noises} \end{bmatrix}$$

가

(reference input)

(disturbance),

(tracking error) ,

(control input) .

$T$   $\infty$ -Norm 1 가

### 3.9 $\mu$ Value $\mu$ Tests

$$\Delta \quad (36)$$

$$\Delta = \{diag[\delta_1 I_{r_1}, \dots, \delta_S, \Delta_1, \dots, \Delta_F] : \delta_i \in \mathbf{C}, \Delta_j \in \mathbf{C}^{m_j \times m_j}\} \quad (36)$$

$$G \in \mathbf{C}^{n \times n} \quad \mu \quad (\mu_\Delta(G(j\omega))) \quad (37)$$

$$\mu_\Delta(G) := \frac{1}{\min \{\bar{\sigma}(\Delta) : \Delta \in \Delta, \det(I - G\Delta) = 0\}} \quad (37)$$

$$(36) \quad \Delta \text{가 } s = \dots, F = 0, r_1 = n$$

가 , (parametric uncertainty)

$$\mu_\Delta(M) = \rho(M)$$

Matrix  $G$  Spectral Radius .  $\rho(M)$

Spectral Radius .

$$, (36) \quad \Delta \text{가 } s = 0, F = \dots, m_1 = n ,$$

(unmodeled dynamics)

$$\mu_{\Delta}(M) = \overline{\sigma}(M)$$

$G$

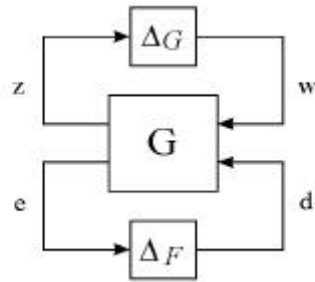


Fig.16 Stability configuration for robust performance

Fig.16  $P \quad \Delta$

$\mu$

$$\Delta_P := \{ \text{diag}[\Delta_G \ \Delta_F] : \Delta_G \in C^{z \times w}, \Delta_F \in C^{e \times d} \}$$

(38)

$\Delta_P$  (extended uncertainty set)

$\Delta_F$  가 (fictitious uncertainty)

$$\mu_{\Delta}(G) = \mu_{\Delta} [F_U\{\Delta, (F_L(P, K))\}]$$

$$= \min_{\substack{K(s) \\ \text{Stabilizing}}} \|F_U(\Delta, (F_L(P, K)))\|_{\mu_{\Delta}} \quad (39)$$

, (39)

$\mu$

Controller

### 3.10 가

1) (Nominal Performance : NP)

$e$   $d$  가 (40)

Nominal Performance . ,  $\Delta_G = 0$

$$\sup \bar{\sigma} (P_{ed}(j\omega)) \leq 1, \forall \omega \quad (40)$$

2) (Robust Stability : RS)

$z$   $w$  가 (41)

Robust Stability

$$\sup \mu_{\Delta_G} (P_{zw}(j\omega)) \leq 1, \forall \omega \quad (41)$$

3) (Robust Performance : RP)

Fig.16 (augmented uncertainty)  $\Delta_P$

$P$  가 (42) Robust

Performance

$$\Delta_P = \begin{bmatrix} \Delta_G & 0 \\ 0 & \Delta_F \end{bmatrix}$$

$$\sup \mu_{\Delta_P} (P(j\omega)) \leq 1, \forall \omega \quad (42)$$

### 3.11 D- K

D Scaling Matrix

K

K  $\mu$  Controller

. D

(39)

D- K

(43- 1) (43- 2)

$$\mu(K) \leq \min_{D \in D} \overline{\sigma}(DKD^{-1}) \quad (43- 1)$$

$$\min_K \left( \min_{D \in D} \|DKD^{-1}\|_{\infty} \right) \quad (43- 2)$$

K Step

Scaling Matrix  $D(s)$

Controller K

$$\min_{K(s)} \|DKD^{-1}\|_{\infty}$$

D Step

$N(K)$

$$\overline{\sigma}(DND^{-1}(j\omega))$$

$D(j\omega)$

$$\mu_{\Delta}(G) = \mu_{\Delta} [F_U \{ \Delta, (F_L(P, K)) \}]$$

## 4

### 4.1. 4

Fig.1 Yaw Motion Side Slip Motion

2 .

### 4.2

(nominal) (44)

$$G_0(s) = \frac{y(s)}{u(s)} = C (sI - A)^{-1} B + D$$

(44)

$$G_0(s) \quad (45)$$

가 .

$$G_0(s) = \begin{bmatrix} \frac{0.0009s^2 + 0.5384s + 2.0368}{s^2 + 629.3s + 1510.8} & \frac{-40.9s^2 - 25563.2s - 21100.6}{s^2 + 629.3s + 1510.8} \\ \frac{0.00002s + 0.015}{s^2 + 629.3s + 1510.8} & \frac{1.3s + 800.2}{s^2 + 629.3s + 1510.8} \end{bmatrix} \quad (45)$$

### 4.3

Fig.17 (  $G_0$  ) (  $\Delta$  ) (  $K$  )

Table 1 .

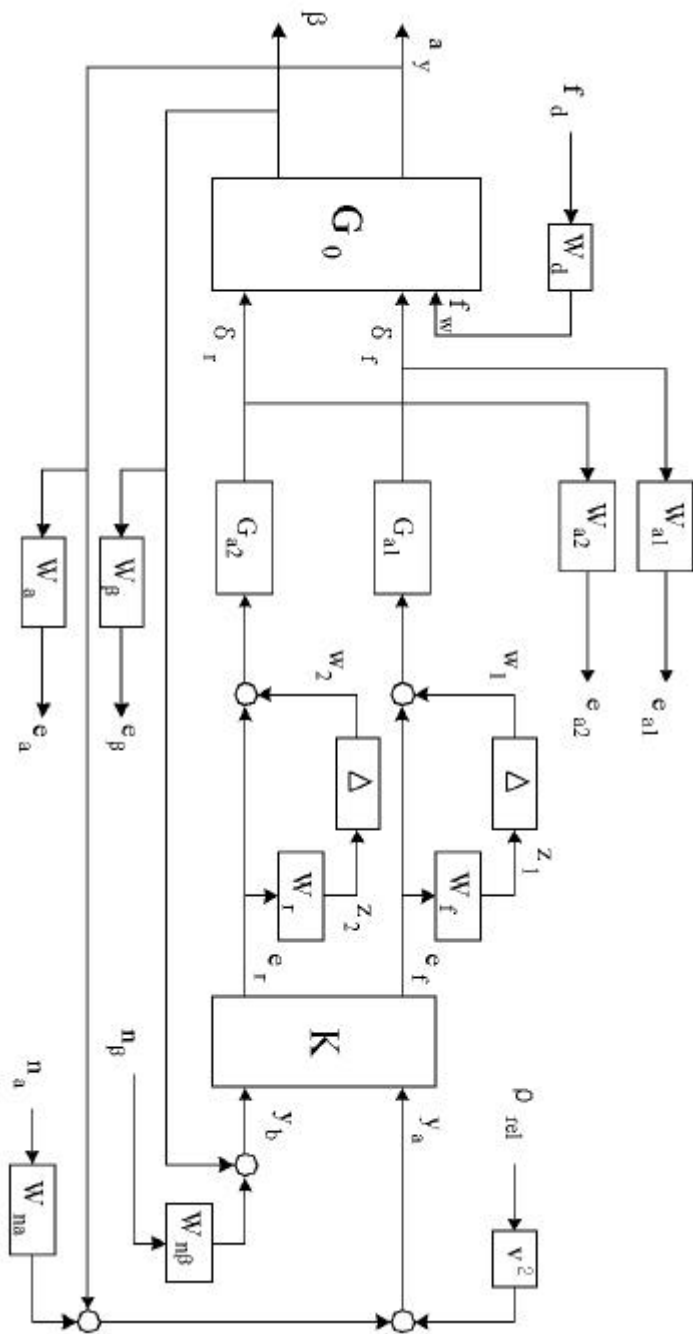


Fig.17 General feedback structure for lateral vehicle control system

Table 1: Description of block diagram for figure.17

	$f_d$
	$\delta_f$
	$\delta_r$
가	$a_y$
	$\beta$
	$G_{a1} , G_{a2}$
가	$W_f , W_r$
가	$W_d$
가	$W_{a1} , W_{a2}$
feedback 가	$W_\beta , W_a$
(noise) 가	$W_{n\beta} , W_{na}$
	$\rho_{rel}$
가	$V^2$

$$H_\infty \quad \mu$$

(A)

(K)

Fig.18



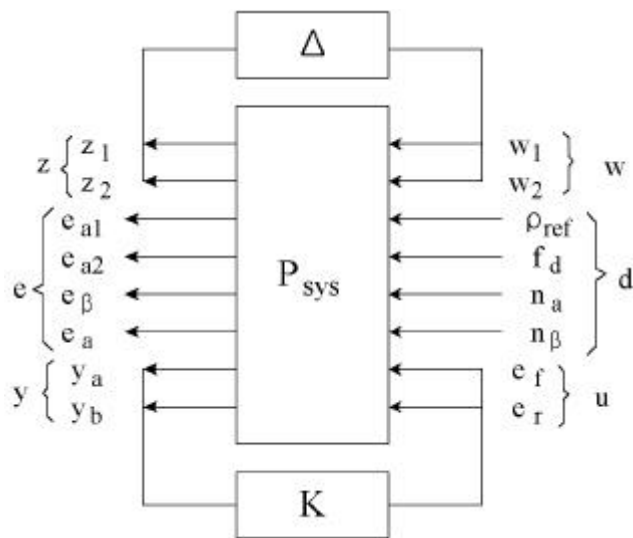


Fig.18 Schematic block diagram of augmented vehicle plant

#### 4.4 Open Loop Interconnection

Fig.18

Δ

, Controller  $K$

.

#### 4.5 LFT (Linear Fractional Transformation)

1) Lower LFT

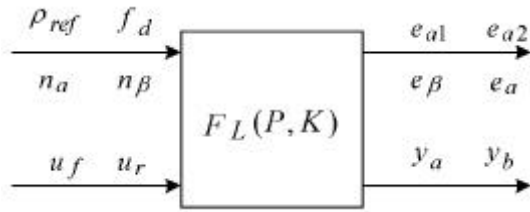


Fig.19 Control system with lower LFT

$$(33-1) \quad \begin{matrix} (P) & (K) \\ d & e \end{matrix} \quad \text{Fig.19} \quad (46-1)$$

$$F_L(P, K) = P_{11} + P_{12}K(I - P_{22}K)^{-1}P_{21} \quad (46-1)$$

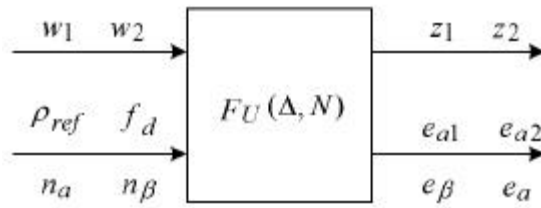


Fig.20 Control system with upper LFT

2) Upper LFT

$$(33-2) \quad \begin{matrix} \text{Lower LFT} & \text{Fig.20} \\ \text{Upper LFT} & (46-2) \end{matrix}$$

$$F_U(\Delta, N) = N_{22} + N_{21}\Delta(I - N_{11}\Delta)^{-1}N_{12} \quad (46-2)$$

## 4.6 가

가

가

가 (47)

$$W_d(s) = \frac{0.1s + 0.8}{0.5s + 1} \quad (47)$$

(noise)

가

(48)

$$W_{na}(s) = W_{n\beta}(s) = \frac{0.25(s + 1)}{0.05s + 1} \quad (48)$$

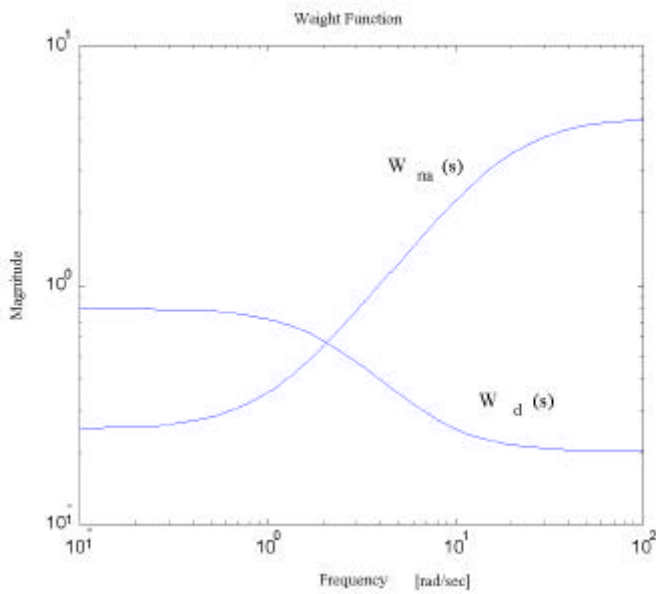


Fig.21 Weight function for  $W_{na}(s)$  and  $W_d(s)$

Fig.21 (47) (48) 가

(49)

$$G_{a1}(s) = G_{a2}(s) = \frac{0.001s + 1}{0.1s + 1} \quad (49)$$

, Control

(50), (51)

가

$$W_{a1} = \frac{57.3}{20} \quad (50)$$

$$W_{a2} = \frac{57.3}{15} \quad (51)$$

(52), (53) 가

$$W_{\beta} = \frac{57.3}{5} \quad (52)$$

$$W_a = \frac{1}{6} \quad (53)$$

$W_{\beta}$  Side Slip Angle  $5^{\circ}$   $W_a$

가 가  $6m/s^2$  가 가

가

## 4.7

가

가

$$W_{p1}(s) = W_{p2}(s) = \frac{0.2s + 2.5}{0.4s + 1.5} \quad (54)$$

(55)

가

가

$$W_{actuator} := \begin{bmatrix} W_{p1}(s) & 0 \\ 0 & W_{p2}(s) \end{bmatrix} = \begin{bmatrix} \frac{0.2s+2.5}{0.4s+1.5} & 0 \\ 0 & \frac{0.2s+2.5}{0.4s+1.5} \end{bmatrix} \quad (55)$$

Fig.22 (54) 가 .

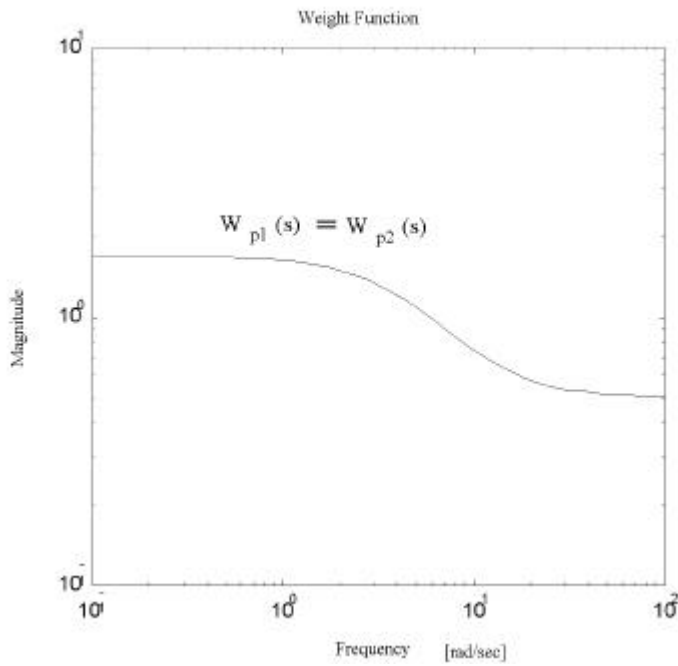


Fig.22 Weight function for  $W_{p1}(s)$  and  $W_{p2}(s)$

(55)

(56)

$$G_{real} := \{G_{nominal}(I + \Delta_G W_{actuator}) : \|\Delta_G\|_\infty \leq 1\} \quad (56)$$

(57)

$$G_{real} := \left\{ G_0(s) \left( I_2 + \Delta_G(s) \begin{bmatrix} \frac{0.2s+2.5}{0.4s+1.5} & 0 \\ 0 & \frac{0.2s+2.5}{0.4s+1.5} \end{bmatrix} \right) \right\} \quad (57)$$

# 5

## 5.1

Table 2

Table 2: Typical vehicle data

M	1500 [kg]
$I_{zz}$	2400 [kg · m <sup>2</sup> ]
$C_{cf}$	55000 [N/rad]
$C_{cr}$	45000 [N/rad]
$l_f, l_r$	1.2, 1.8 [m]
$l_w$	0.4 [m]
V	100 [km/h]

## 5.2 $H_\infty$ Controller가 $\mu$

$H_\infty$  Controller가

Table 3

Table 3: Closed-loop poles of  $H_\infty$  controller

(Eigenvalues)					
- 3887.52	- 522.84	- 312.24	- 70.96	- 19.04	- 7.54
- 3.65	- 4.04	- 2.00			

가

$H_\infty$  Controller



(58)  $\gamma_{opt} = 5.1634$

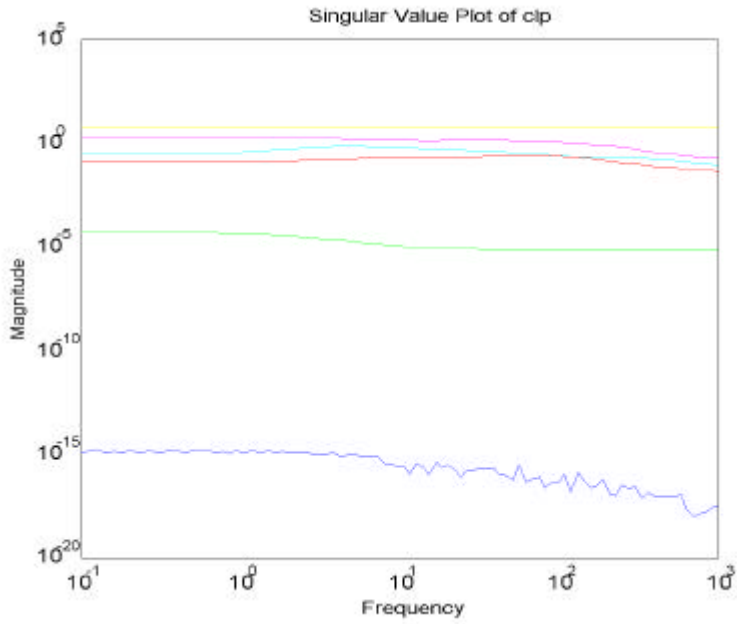


Fig.23 Singular value plot of close-loop system

Fig.23 Open Loop Interconnection

(singular value)

Fig. 24  $H_\infty$  Controller Open Loop Interconnection

Lower LFT

$H_\infty$  Controller

(robust stability)

(nominal performance)

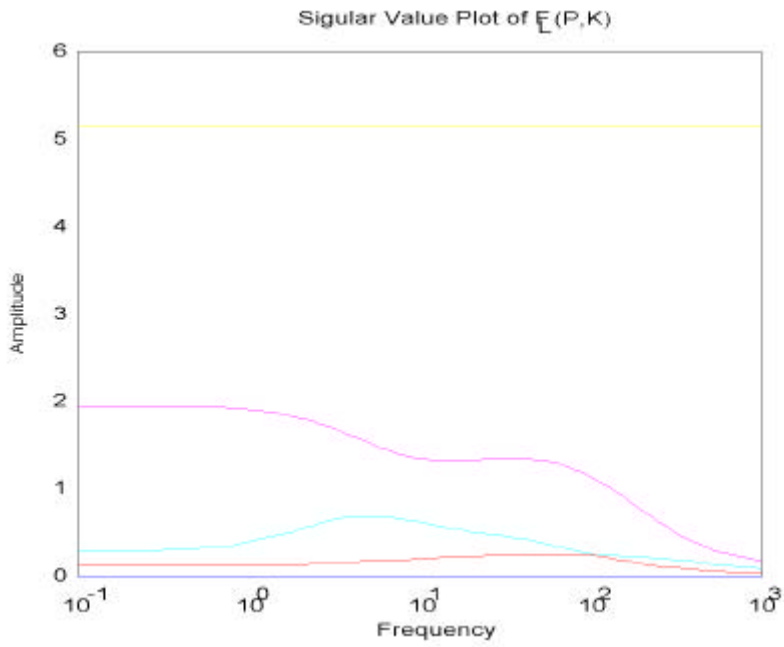


Fig.24 Singular value plot of lower LFT

Fig.25

가

Fig.26

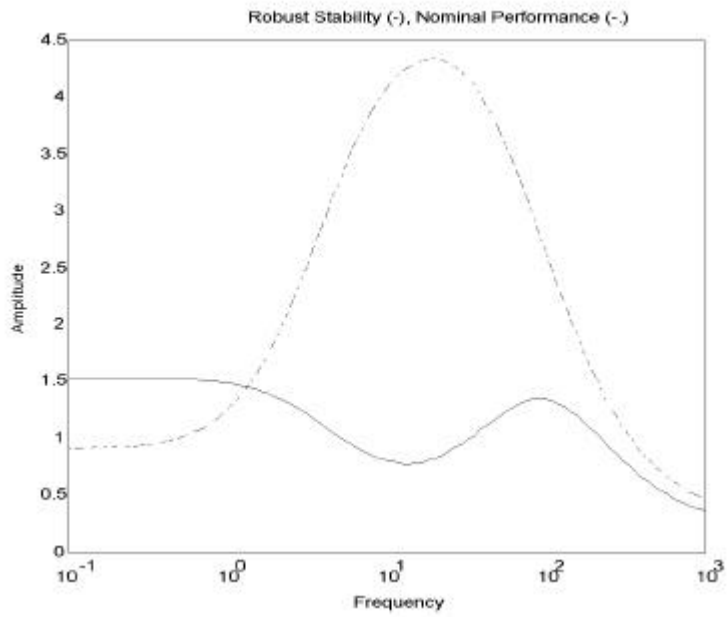


Fig.25 Robust stability and nominal performance

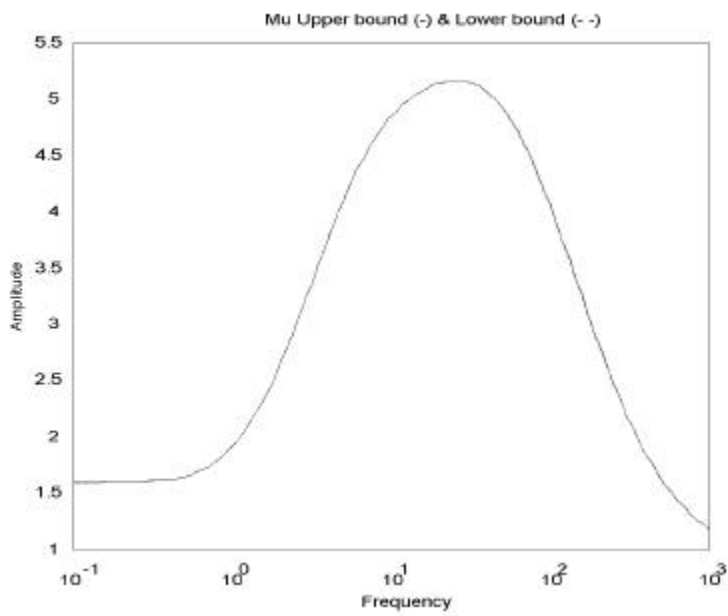


Fig.26 Structured singular values( $\mu$ ) with upper and lower bounds

Robust Performance (RP)

NP

RS

$D-K$

$D-K$

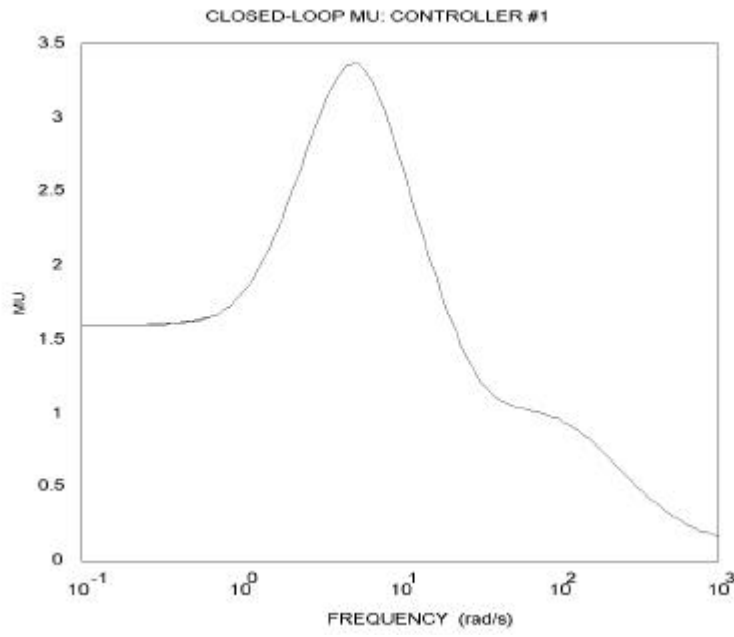


Fig.27  $\mu$  value for controller #1

Fig.27  $D-K$

$\mu$

가

, NP

RS

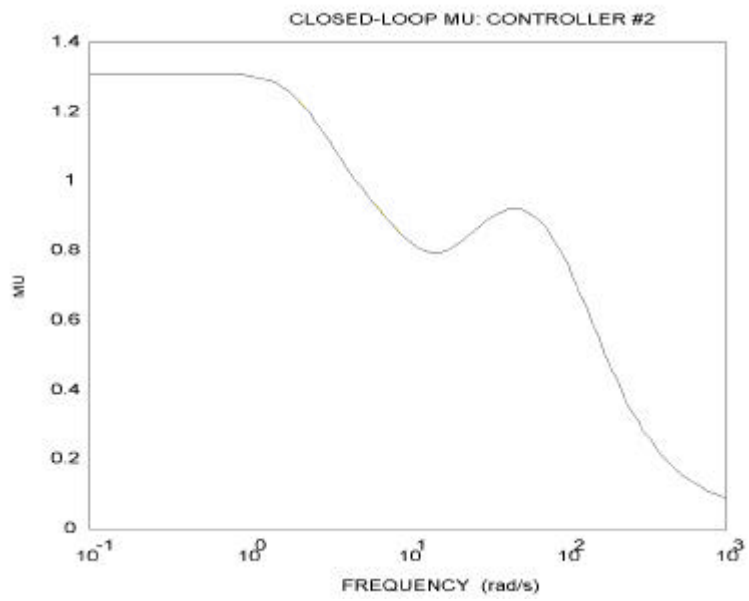


Fig.28  $\mu$  value for controller #2

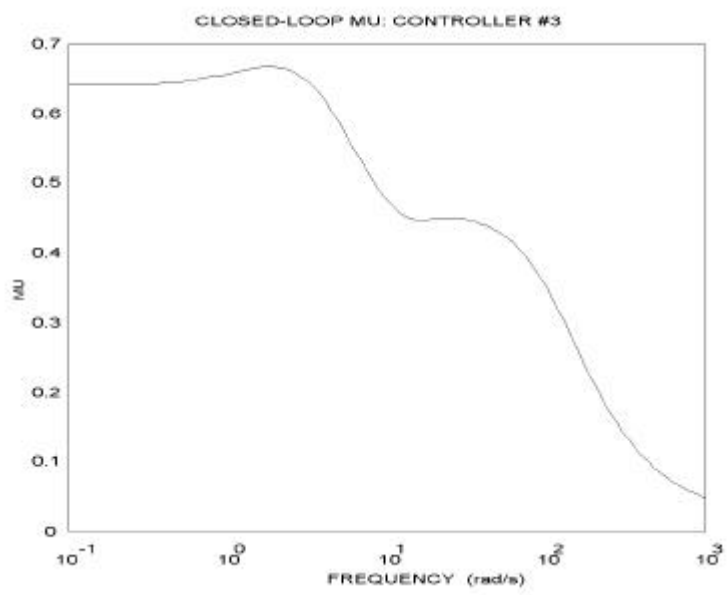


Fig.29  $\mu$  value for controller #3

Fig.28  $D-K$   $\mu$

가  
, NP RS

Fig.29  $D-K$   $\mu$

가 (39)

$\mu$  Controller

NP, RS, RP

Fig.30  $D-K$   $K$

$H_\infty$  Controller  $\mu$

Controller Fig.25 Fig.30

$\mu$  Controller가

가 .

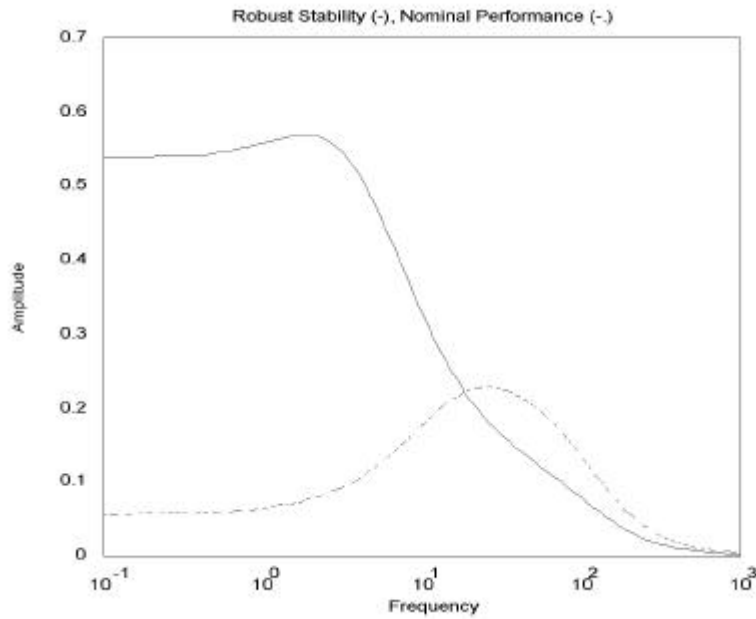


Fig.30 Robust stability and nominal performance for controller #3

Table 4

Table 4: D- K iteration summary

Iteration Summary			
Iteration #	1	2	3
Controller Order	9	13	17
Total D- Scale Order	0	4	8
Gamma Acheived	5.584	1.697	0.954
Peak mu- Value	3.370	1.307	0.667

Table 5

Table 5: Close-loop poles of  $\mu$  controller

(Eigenvalues)				
- 522.10	- 148.35	- 68.84	- 16.64 + 0.01i	- 16.64 - 0.01i
- 13.55	- 9.11	- 5.49	- 2.86 + 0.01i	- 2.86 - 0.01i
- 1.01	- 1.58	- 64.76	- 64.76	- 1.99
- 2.88	- 2.88			

Table 5

*D- K*

### 5.3

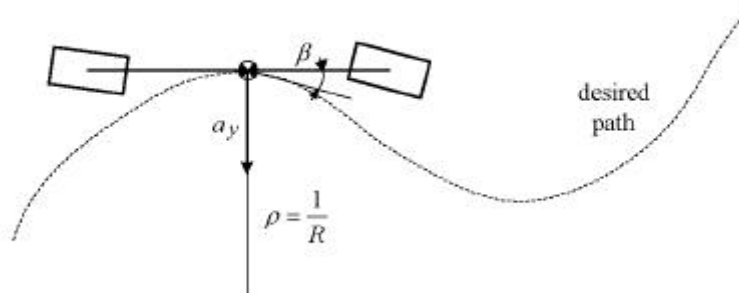


Fig.31 Vehicle maneuver along the desire road path



Table 6: Desired radius (R) of road path in time-domain simulation

( [sec])	0 2	2 4	4 7	7 10	10
R ( [m])	$\infty$	50	-60	70	$\infty$

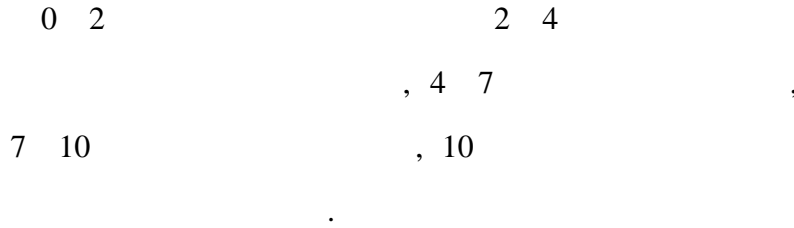


Fig.31

가

Table 6

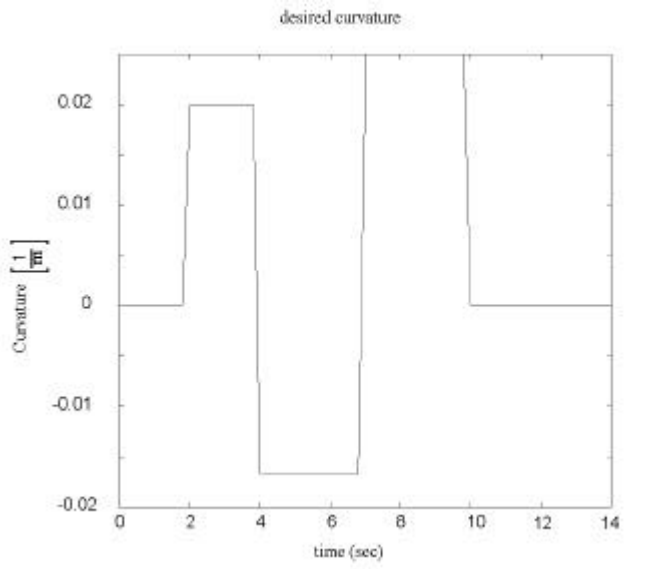


Fig.32 Road curvature input

Fig.32

가

가

Fig.33

Fig.34

Fig.33

가

가

가

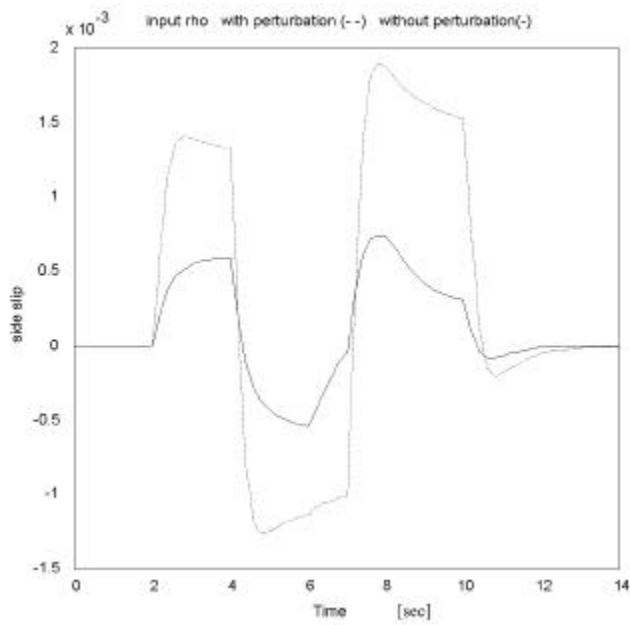


Fig.33 Time-domain response for vehicle side slip

Fig. 34

가

가 (lateral acceleration)

(perturbation)

가

가

가 가

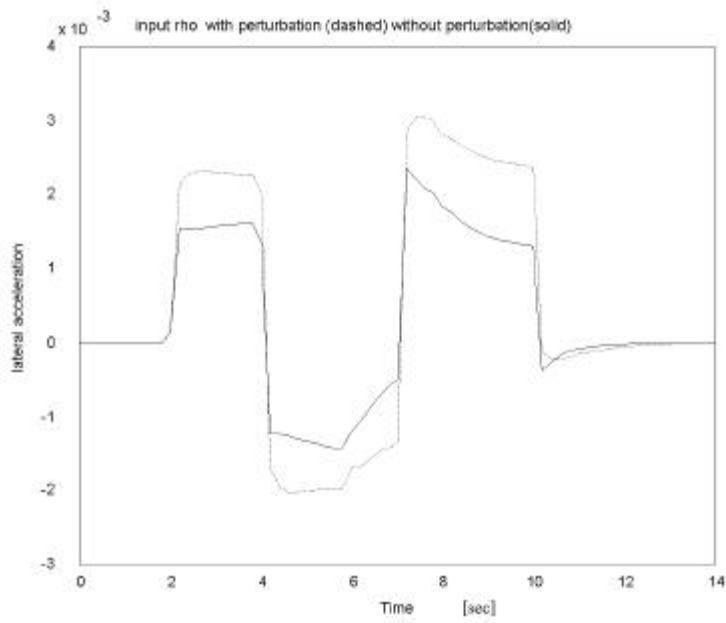


Fig.34 Time-domain response for vehicle lateral acceleration

# 6

가

가

,  
가 .

$H_\infty$   $\mu$

$H_\infty$   $\mu$

가

,  
가 .

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