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低速 機關 燃料噴射系統

Simulation of the Fuel Injection System
in Low Speed Diesel Engines

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ABSTRACT

Diesel invented the compression-ignition engine in 1892. Since that time these engines have continued to develop as our knowledge of engine processes has increased. So, nowadays they play a dominant role in the fields of automobiles, ships and some prime movers. But now worldwide concerns with global climate and environmental protection changed the trend of diesel engine researches to solve the problems how to reduce the pollutant emissions from diesel engines to meet the restrict emission regulations by the IMO(International Maritime Organization).

There are several method that reduce the emission. First of all, the fuel injection system of a diesel engine has taken more important place in understanding of diesel combustion process with combustion chamber, and has taken one of the most important part to prevent environmental pollution by exhaust gas from diesel engine. From this point of view, many investigations have been carried out to solve this problem, such as adopting higher injection pressure and shortening the injection duration by the higher injection rate, etc. Owing to this effort there are considerable improvement to solve pending issues.

But these researches are mainly on the high speed diesel engine or spark ignition engine, therefore it is worth while to study the

low speed diesel engine for ship's use to compare the results which was well known for general trend by the previous researches.

In this study the analysis was carried out by simplifying and modeling the injection phenomena and dividing into three parts comprising of fuel injection pump, high pressure pipe and fuel injection nozzle in the fuel injection system of a low speed diesel engine. A computer simulation model was developed using the Runge-kutta method to solve the equations for each part (fuel injection pump, high pressure pipe and fuel injection nozzle) and the method of characteristics to analyze the unsteady flow in the fuel injection system considering cavitation and variation of fuel density and bulk modulus. Applied was the constant pressure condition at the nodes in the high pressure pipe.

Comparison was commenced between the calculated data and experimental data of pressure and injection quantity at the fuel oil distributor in fuel injection system for the training ship Hanara. In the work presented here, the results of a new model which was developed about low speed diesel engine was similar trend to earlier works in the high speed engines. Simulation results about the effect of the high pressure pipe diameter, length, sac volume and efflux coefficient was also analyzed.

NOMENCLATURE

- A : Area (cm²)
- B_1 : First section of Pipe.
- B_2 : Last section of Pipe.
- B_3 : First section of Pipe.
- B_4 : Last section of Pipe.
- C : Efflux coefficient
- D : Diameter of high pressure pipe (cm)
- F : Initial force (Kgf)
- K : Bulk modulus of fuel oil (Kgf/cm²)
- k : Stiffness of spring (Kgf/cm)
- M : Mass (Kgf · s²/cm)
- P : Pressure (Kgf/cm²)
- Q : Volumetric flow rate (cm³/s)
- Re : Reynolds number
- U : Velocity (cm/s)
- V : Volume (cm³)
- Y : Lift (cm)
- a : Velocity of wave propagation (cm/s)
- f : Darcy-Weisbach friction factor (s⁻¹)

- g : Gravitational constant (cm/s^2)
 t : Time (s)
 x : position
 ρ : Fluid density ($\text{Kgf} \cdot \text{s}^2/\text{m}^4$)
 θ : Angle (deg.)
 λ : Damping coefficient ($\text{Kgf} \cdot \text{s}/\text{cm}^2$)

Subscript

- A,B,C,R,S,P : Points of x - t Plane
 N : Number of grid in pipe
 NS : $N + 1$
 cyl : Combustion chamber
 e : Cavitation
 sp : Spill port
 h : Nozzle hole
 l : High pressure pipe
 liq : Liquid
 n : Needle
 no : Area of nozzle valve opening
 u : Plunger

sc : Sac chamber

vap : Vapor

,j : Section of j grid at I time for pipe

P ,j : Section of j grid at *P* time for pipe (*P* = + Δt)

,j : Section of j grid at time for pipe

P ,j : Section of j grid at *P* time for pipe (*P* = + Δt)

, Soot
 . Soot
 , NOx
 .
 NOx Soot (Trade off)
 가
 . NOx Soot
 , ,
 가
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 , 2
 , 가
 ,
 ,
 가 .

EMS(Engine Monitoring System)

4

-

,

4)

,

,

5),6)

.

2

2.1

Fig.2- 1 ,

(Pipe)

(Pipe)

2.2

가

가 .

(1) 1 , .

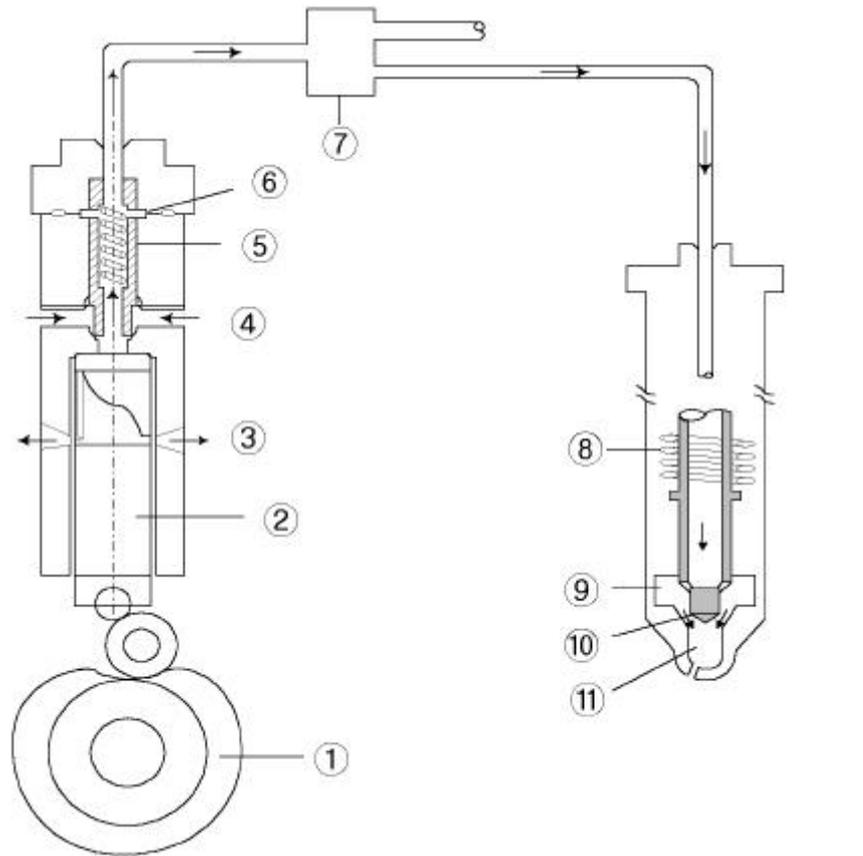
(2)

(3)

(4)

(5)

(Spill port)



- | | |
|---------------------|-------------------------|
| 1. Fuel cam | 7. Oil distributor |
| 2. Plunger | 8. Nozzle spring |
| 3. Spill port | 9. Nozzle chamber |
| 4. Intake port | 10. Nozzle needle valve |
| 5. Slide valve | 11. Sac chamber |
| 6. Delivery chamber | |

Fig. 2-1 Schematic diagram of fuel injection system

2.3

2.3.1

$$A_u U_u$$

$$C_{sp} A_{sp} \sqrt{\frac{2}{\rho} [P_u - P_{sp}]}$$

$$A_l U_{l,j}$$

$$\frac{dP_u}{dt} = \frac{K_u}{V_u} \left(A_u U_u - C_{sp} A_{sp} \sqrt{\frac{2}{\rho} [P_u - P_{sp}]} - A_l U_{l,j} \right) \quad (2.1)$$

2.3.2

(Sac Chamber)

가 가

가 .

(1)

$$A_l U_{l,j}$$

$$A_n U_n$$

$$C_n A_{no} \sqrt{\frac{2}{\rho} [P_n - P_{sc}]}$$

$$\frac{dP_n}{dt} = \frac{K_n}{V_n} \left(A_{l,j} U_{l,j} - A_n U_n - C_n A_{no} \sqrt{\frac{2}{\rho} [P_n - P_{sc}]} \right) \quad (2.2)$$

(2)

$$\begin{aligned} & M_n \\ & (A_n - A_{sc})P_n + A_{sc}P_{sc} - F_n - k_n L_n - \lambda_n U_n \end{aligned}$$

$$\frac{dU_n}{dt} = \frac{1}{M_n} \left((A_n - A_{sc})P_n + A_{sc}P_{sc} - F_n - k_n L_n - \lambda_n U_n \right) \quad (2.3)$$

(3) (Sac Chamber)

$$\frac{dP_{sc}}{dt} = \frac{K_{sc}}{V_{sc}} \left(C_n A_{no} \sqrt{\frac{2}{\rho} [P_n - P_{sc}]} - A_h C_h \sqrt{\frac{2}{\rho} [P_{sc} - P_{cyl}]} \right) \quad (2.4)$$

2.3.3

1 가 ,

, 7) .

(1)

$$L_1 = \frac{p_x}{\rho} + VV_x + V_t + \frac{fV|V|}{2D} = 0 \quad (2.5)$$

(2)

$$L_2 = Vp_x + p_t + \rho a^2 V_x = 0 \quad (2.6)$$

$$f \quad \frac{fV^2}{2D}$$

$$\frac{f}{2D} V|V|$$

$$L = L_1 + \lambda L_2 = \lambda \left[p_x \left(V + \frac{1}{\lambda \rho} \right) + p_t \right] + \left[V_x (V + \lambda \rho a^2) + V_t \right] + \frac{fV|V|}{2D} = 0 \quad (2.7)$$

$$\frac{dx}{dt} = V + \frac{1}{\lambda \rho} = V + \lambda \rho a^2 \quad (2.8)$$

(2.8) λ

$$\lambda = \frac{1}{\rho a} \quad , \quad (2.9)$$

$$(2.8) \quad \frac{dx}{dt} = V \pm a \mathcal{A} \quad . \quad (2.10)$$

(2.7)

$$\frac{dV}{dt} + \frac{1}{\rho a} \frac{dp}{dt} + \frac{fV|V|}{2D} = 0 \quad (2.11)$$

(2.10) (2.11) 2

$$\frac{dV}{dt} + \frac{1}{\rho a} \frac{dp}{dt} + \frac{fV|V|}{2D} = 0 \quad (2.12)$$

$$\frac{dx}{dt} = V + a \quad (2.13)$$

$$\frac{dV}{dt} - \frac{1}{\rho a} \frac{dp}{dt} + \frac{fV|V|}{2D} = 0 \quad (2.14)$$

$$\frac{dx}{dt} = V - a \quad (2.15)$$

(2.12) (2.14) $x-t$ plane , (2.12)

(2.13) , (2.14) (2.15) .

(V) (x) (t) $x-t$

(V)

(a) 가 .

Fig.2-2 $\frac{\Delta t}{\Delta x}$

$$\Delta t(V + a) \leq \Delta x$$

Fig.2-2 P , $t = t_0$

A, B, C A, B B, C R

S $t = t_0 + \Delta t$

(2.12) (2.14) .

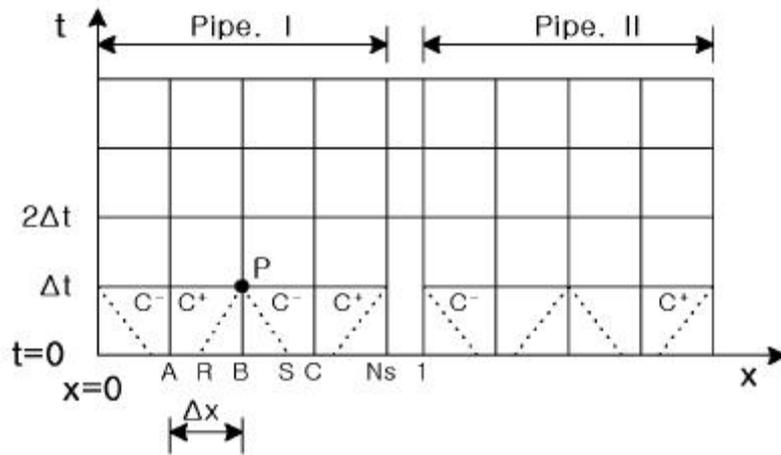


Fig. 2- 2 Characteristic lines in the xt plane.

Fig.2- 3

(Constant pressure)

7)

$$H = H_{,NS} = H_{,1}$$

$$Q_{,NS} = Q_{,1} \times 2$$

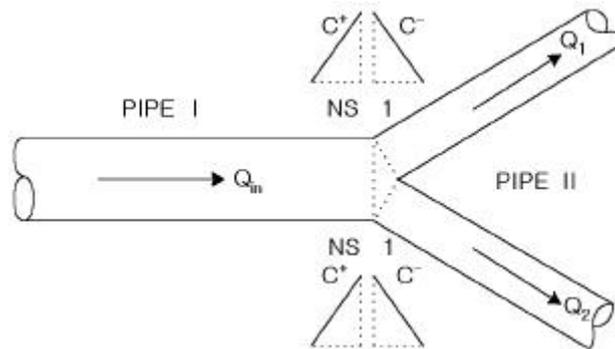


Fig. 2- 3 Schematic of a branching connection

2.3.4

$$(1) \quad (B1)$$

$$\text{Fig.2-4} \quad B1 \quad (2.17)$$

$$(2.16) \quad P_{P,1} \quad (2.18) \quad .8)$$

$$P_{P,1} = a\rho(V_{P,1} - V_S + \frac{1}{a\rho}P_S + \frac{f dt}{2D}V_S|V_S|) \quad (2.16)$$

$$V_{P,1} = \sqrt{\frac{2(P_u - P_{P,1})}{\rho}} \quad (2.17)$$

$$V_{P,1} = V_S + \frac{1}{a\rho}(P_{P,1} - P_S) - \frac{f dt}{2D}V_S|V_S| \quad (2.18)$$

$$V_S = \frac{V_{I,1} - a \frac{dt}{dx}(V_{I,1} - V_{I,2})}{1 - \frac{dt}{dx}(V_{I,1} - V_{I,2})} \quad (2.19)$$

$$P_S = P_{I,1} + \frac{dt}{dx}(V_S - a)(P_{I,1} - P_{I,2}) \quad (2.20)$$

$$(2) \quad (B2)$$

$$B2 \quad (2.21), (2.22)$$

$$(2.23) \quad .$$

$$P_{P,NS} = a\rho(V_R + \frac{1}{a\rho}P_R - \frac{f dt}{2D}V_R|V_R| - V_{P,NS}) \quad (2.21)$$

$$V_{P,NS} = \sqrt{\frac{2(P_{P,NS} - P_{,NS})}{\rho}} \quad (2.22)$$

$$V_{P, NS} = V_R + \frac{1}{a\rho} P_{R-} - \frac{f dt}{2D} V_R |V_R| - \frac{1}{a\rho} P_{P, NS} \quad (2.23)$$

$$V_R = \frac{V_{I, NS} - a \frac{dt}{dx} (V_{I, NS} - V_{I, N})}{1 + \frac{dt}{dx} (V_{I, NS} - V_{I, N})} \quad (2.24)$$

$$P_{R-} = P_{I, NS} - \frac{dt}{dx} (V_R + a)(P_{I, NS} - P_{I, N}) \quad (2.25)$$

(3) (B3)

B3 (Constant Pressure)

$$P_{P, 1} = P_{P, NS} \quad (2.26)$$

$$Q_{P, 1} = Q_{P, NS} / 2 \quad (2.27)$$

(4) (B4)

Fig.2-4 B4 (2.29)

$$(2.28) \quad P_{P, 1}$$

() (2.30)

8).

$$P_{P, NS} = a\rho(V_R + \frac{1}{a\rho} P_{R-} - \frac{f dt}{2D} V_R |V_R|) \quad (2.28)$$

$$V_{P, NS} = \sqrt{\frac{2(P_{P, NS} - P_N)}{\rho}} \quad (2.29)$$

$$V_{P, NS} = V_R + \frac{1}{a\rho} P_{R-} - \frac{f dt}{2D} V_R |V_R| - \frac{1}{a\rho} P_{P, NS} \quad (2.30)$$

$$V_R = \frac{V_{,NS} - a \frac{dt}{dx} (V_{,NS} - V_{,N})}{1 + \frac{dt}{dx} (V_{,NS} - V_{,N})} \quad (2.31)$$

$$P_R = P_{,NS} - \frac{dt}{dx} (V_R + a)(P_{,NS} - P_{,N}) \quad (2.32)$$

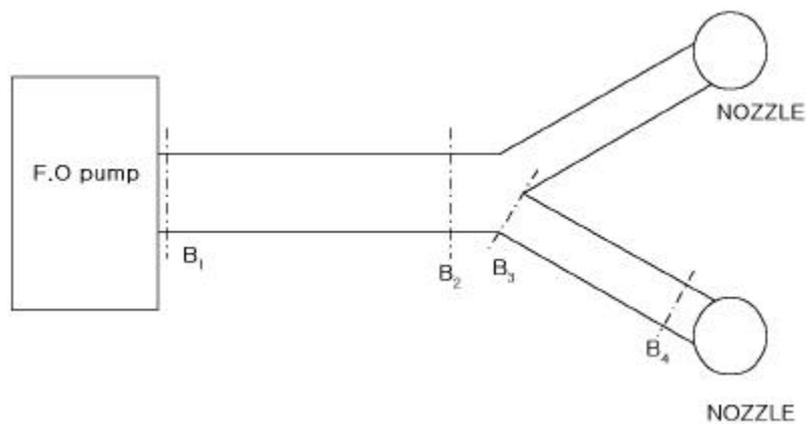


Fig. 2-4 Branching system with fuel pump and nozzle.

2.3.5

(1) ,

$$\rho = \rho_0(1 + aP - bP^2) \quad (2.33)$$

$$K = -V \frac{\partial P}{\partial V} = \frac{1 + aP - bP^2}{a - 2bP} \quad (2.34)$$

ρ_0 , a, b

R.S. Dow C.E. Fink 9) .

(2)

f

Darch- Weisbach

$$: f = \frac{64}{Re} \quad (2.35)$$

$$: f = 0.00019064 Re^{0.64378} \quad (2.36)$$

$$: f = 0.3164 Re^{-0.25} \quad (2.37)$$

(3)

(Cavity)

ρ_e

K_e

$$\rho_e = \rho_i + \frac{\Delta M}{V} \quad (2.38)$$

$$K_e = \frac{K_{vap}}{1 + [(K_{vap} - K_{liq})/K_{liq}]/VL} \quad (2.39)$$

$$VL = \frac{\rho_{liq} - \rho_e}{\rho_{liq} - \rho_{vap}} \quad (2.40)$$

$\rho_i \Delta t$, $\Delta M \Delta t$

가 . K_e

$$VL \quad \rho_{vap} \leq \rho_e \leq \rho_{liq}$$

VL . VL 가

$$VL \geq 1 \quad VL = 1, \rho_e = \rho_{liq}$$

$$VL < 0 \quad VL = 0, \rho_e = \rho_{vap}$$

(4) (Spill port)

1)

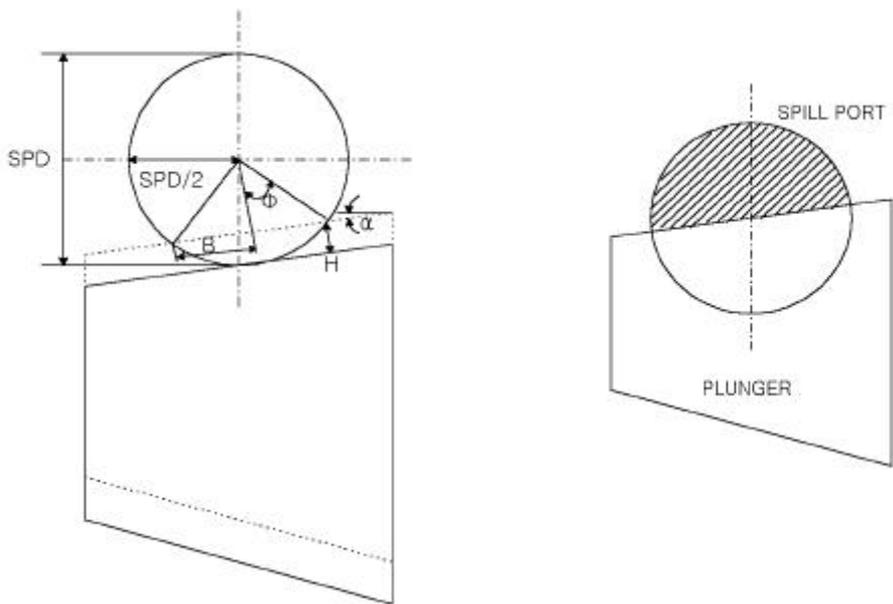


Fig. 2- 5 The area of spill port closing

H (2.44) (XLP)

RPM (SPC)

H 10)

$$H = (XLP - SPC) \cdot \cos \phi \quad (2.41)$$

$$\cos \phi = \left(\frac{SPD}{2} - H \right) / \left(\frac{SPD}{2} \right) \quad (2.42)$$

$$B = \frac{SPD}{2} \sin \phi \quad (2.43)$$

$$H - SPD/2 = 0 \quad (2.42) \quad (2.43)$$

$$A_{sp} = \frac{\pi \cdot \phi}{4} \cdot SPD^2 + B \cdot \frac{SPD}{2} \cos \phi \quad , \quad (2.44)$$

$$H - SPD/2 = 0 \quad SPD \quad (2.45) \quad (2.46)$$

$$\cos \phi = \left(H - \frac{SPD}{2} \right) / \left(\frac{SPD}{2} \right) \quad (2.45)$$

$$B = \left[H - \frac{SPD}{2} \right] \tan \phi \quad (2.46)$$

$$A_{sp} = \frac{\phi}{4} \cdot SPD^2 - B \cdot \frac{SPD}{2} \cos \phi \quad , \quad (2.47)$$

$$H = SPD/2$$

$$A_{sp} = \frac{\pi}{8} \cdot SPD^2 \quad . \quad (2.48)$$

2)

Fig.2- 6

H

(XLP),

(ES)

(2.49)

,

H

10)

$$H = [XLP - (ES + SPD + SPC)] \cos \beta \quad (2.49)$$

$$H = SPD / 2$$

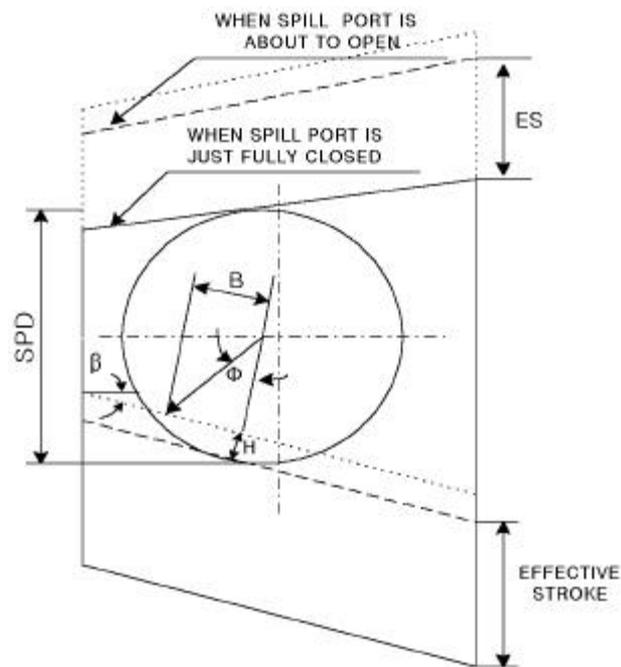
$$A_{sp} = \frac{\phi}{4} \cdot SPD^2 - B \cdot SPD/2 \cos \phi \quad , \quad (2.50)$$

$$H = SPD / 2$$

$$A_{sp} = \frac{\pi - \phi}{4} \cdot SPD^2 + B \cdot SPD/2 \cos \phi \quad , \quad (2.51)$$

$$H = SPD/2$$

$$A_{sp} = \frac{\pi}{8} \cdot SPD^2 \quad . \quad (2.52)$$



**Fig. 2-6 The area of spill port opening
(after the effective stroke)**

(5) Nozzle

Fig.2-7 $A_1()$ $A_2()$

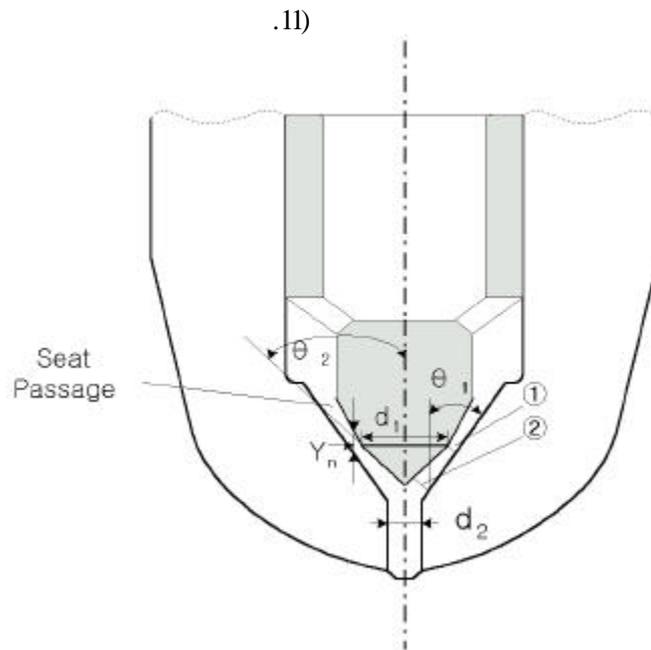


Fig. 2-7 The flow area through nozzle

$$A_1 = \pi (d_1 + 0.5 Y_n \sin(2\theta_1) Y_n \cos \theta_1) \quad (2.53)$$

$$A_2 = \pi [d_2 + \frac{d_1 - d_2}{2} \cos^2(\theta_2) - 0.5(Y_n + \frac{d_1 - d_2}{2} \tan \theta_1) \cdot \sin(2\theta_2)] \cdot [Y_n - \frac{d_1 - d_2}{2} (\tan \theta_1 - \cot \theta_2)] \sin \theta_2 \quad (2.54)$$

(6)

,
.
.
0.6 0.7 12),13)
14) , 0.7,
0.6 ,
0.4 .
8 Kgf/cm 2 30
Kgf/cm 2 .

2.4

4 - ,
2 .

Fig.2- 8 .

(1)

, , ,
.

(2)

,

(3)

(Pipe)

(Pipe)

(4)

4 -

, , .

(5)

$$t = t + \Delta t$$

.

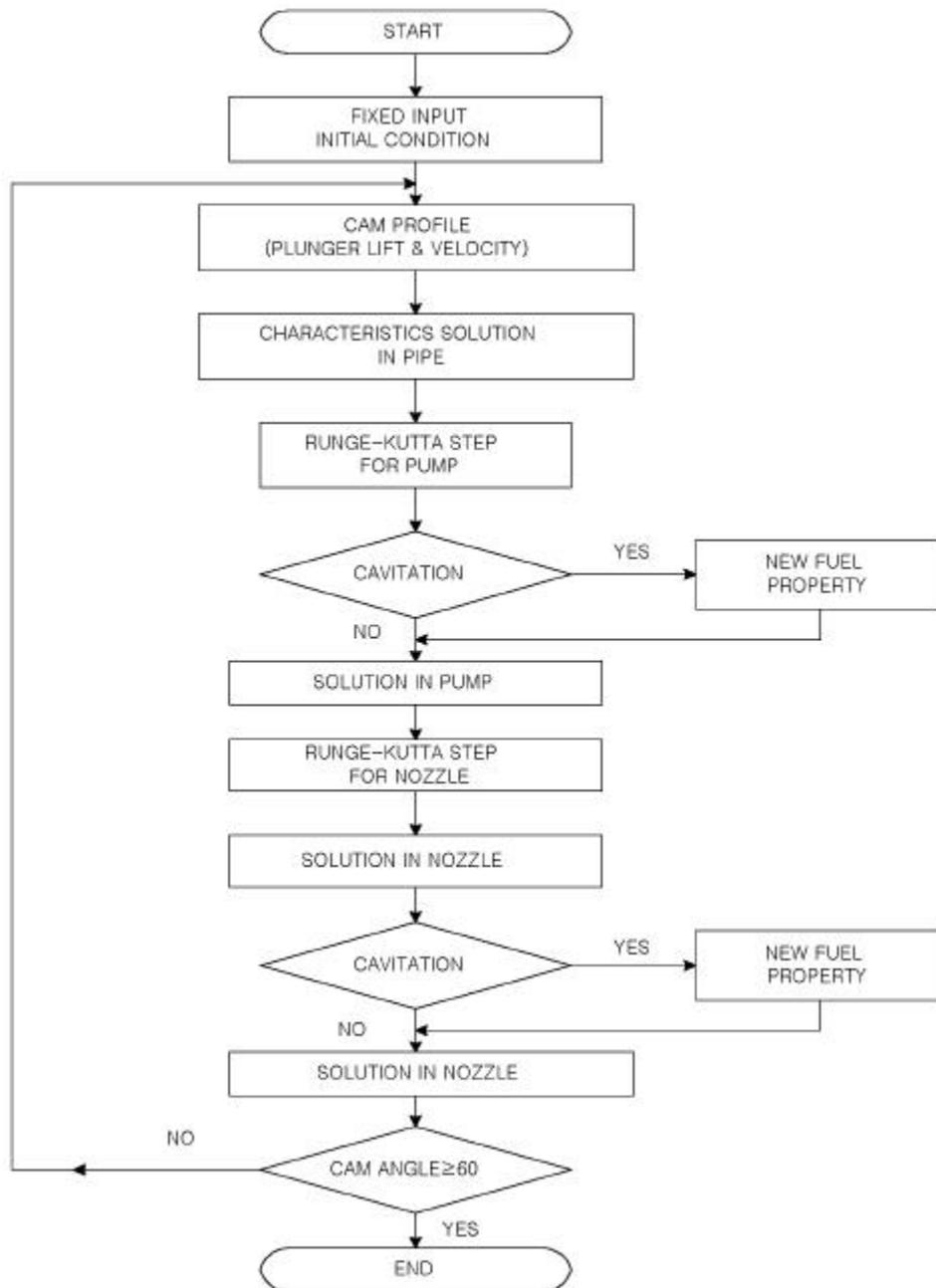


Fig. 2- 8 Flow chart

3

3.1

Data
35MC

M.A.N.- B&W 6L

Table 2- 1 Specification of experimental apparatus

Equipments		Items	Dimension
Fuel pump		Plunger diameter Plunger maximum lift Spill port diameter	28.5 mm 39.12 mm 2.5 mm
High pressure pipe	Pipe.	Length Diameter	716 mm 6.0 mm
	Pipe.	Length Diameter	892 mm 4.0 mm
Fuel injection valve		Diameter of needle large end Diameter of needle small end Needle maximum lift Stiffness of spring Opening pressure of needle valve Sac volume	12.4 mm 6.5 mm 1.6 mm 24.1 Kgf/mm 300 Kgf/cm ² 123 mm ³
Fuel supply pressure : 8 Kgf/cm ²			

3.2

VIT

3.2.1

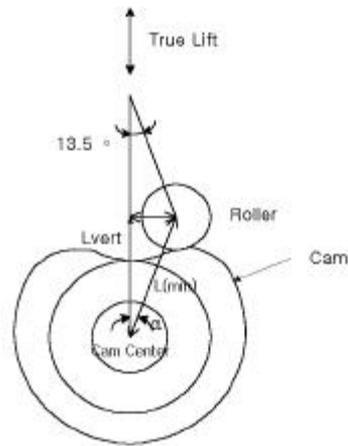


Fig. 3- 1 Fuel cam profile

Fig.3- 1

L

Lvert

(True lift)

Fig.3- 2

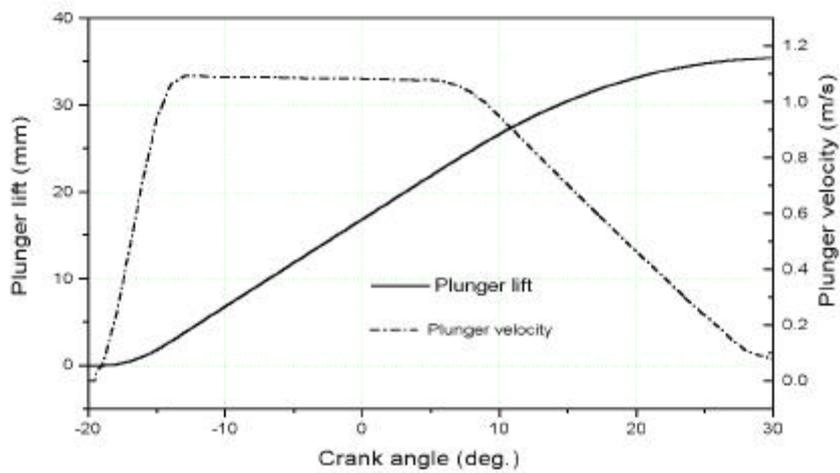


fig. 3- 2 Plunger lift & velocity

3.2.2 VIT

Fig.3-3 (a) (Straight edge type)

, (b) VIT

가 . Fig.3-4

VIT

(Pmax) VIT 가 가

가 ,

VIT 가

, 4500 PS

(Derating) 85%

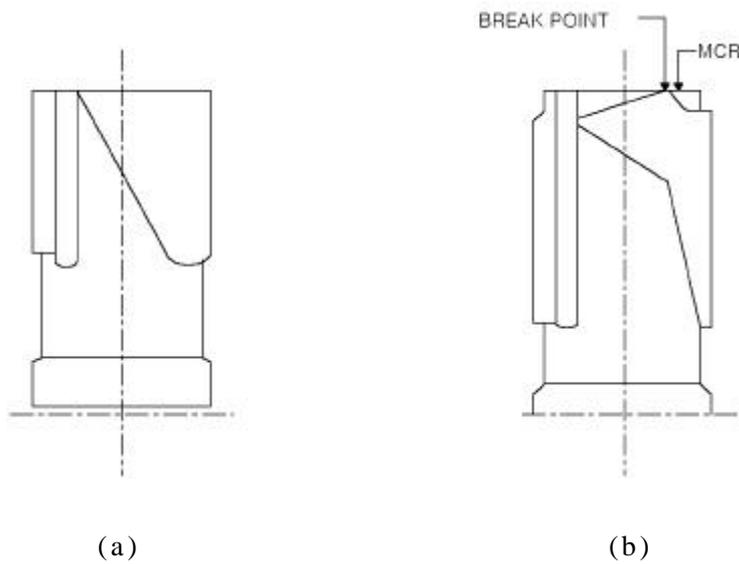


Fig. 3-3 Straight edge Plunger(a) & VIT Plunger(b)

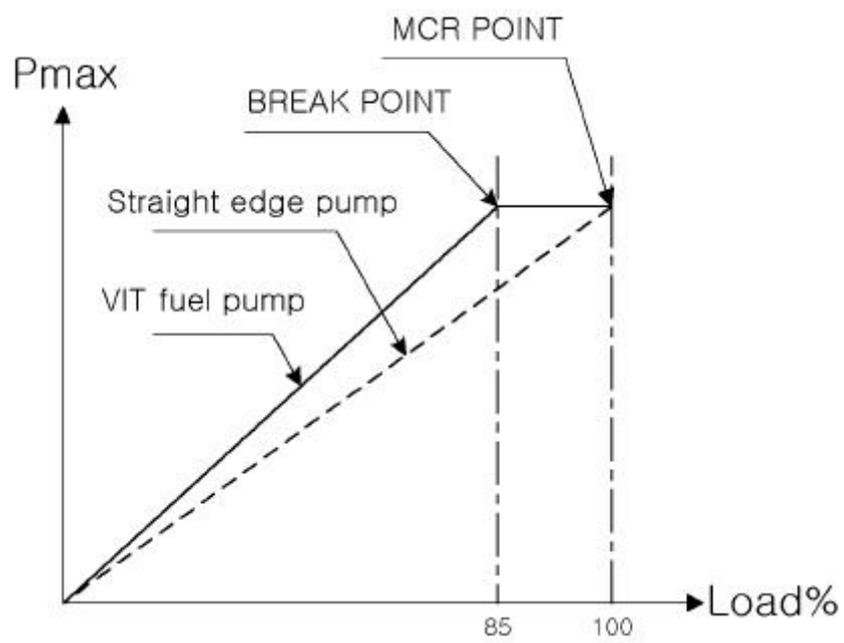


Fig. 3-4 Comparison of Pmax

3.3

3.3.1

EMS(Engine Monitoring System)

가

가

PC

(History)

3.3.2 System

(Piezo- tron)

(Junction box)

(Remote control

box)

가

(Acquisition card)

, TDC

가

가

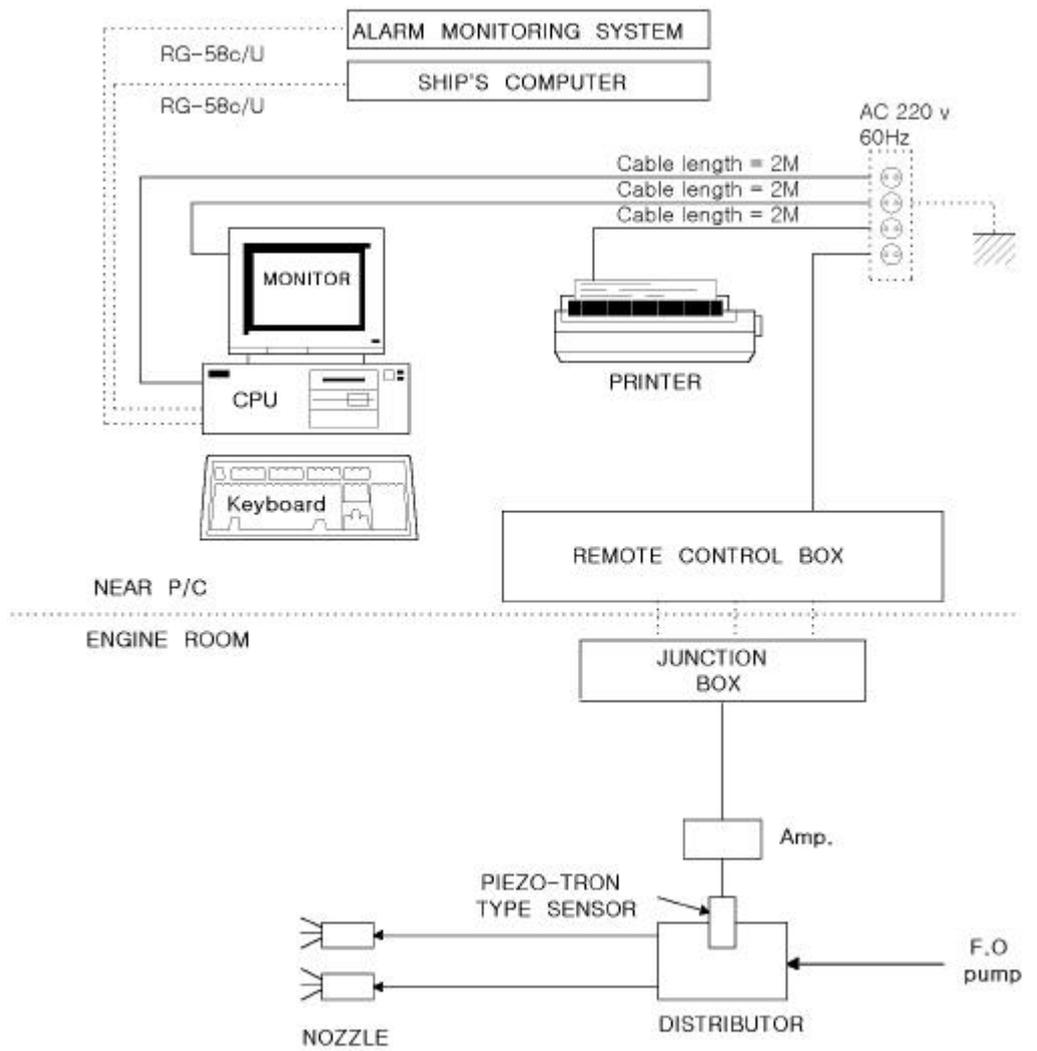


Fig. 3- 5 Engine Monitoring System(Data acquisition system)

4

EMS

(Distributor)

4.1

Fig.4- 1

(Distributor)

RPM

VIT

190 RPM

740 Kgf/cm²

, 160 RPM

,

440

Kgf/cm²

4.2

Fig.4- 2, Fig4- 3, Fig4- 4

RPM

RPM

3%

가

1. ,
 0.01 ° . dt
 가

(精度)

Fig.4- 5 RPM
 가
 (Fuel rack position)

(Break point) 가

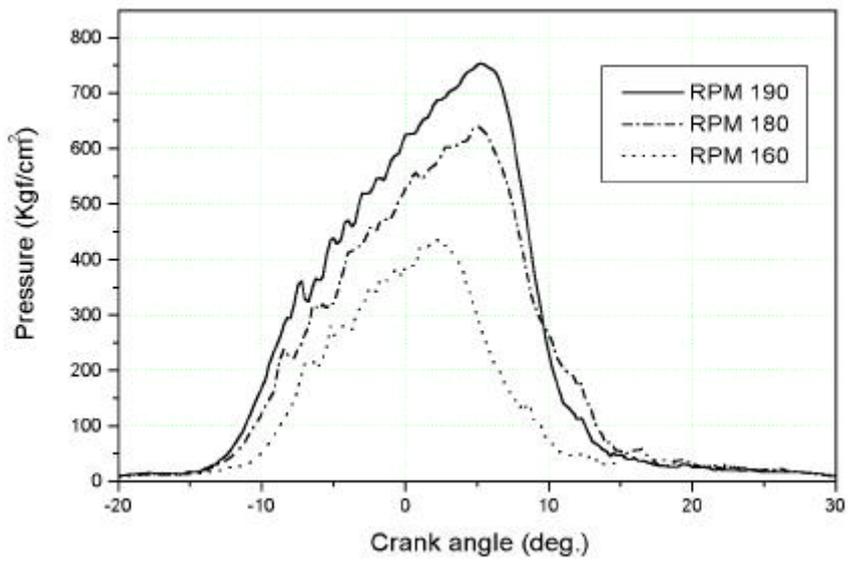


Fig. 4- 1 Experimental results of injection pressure
 (at the distributor)

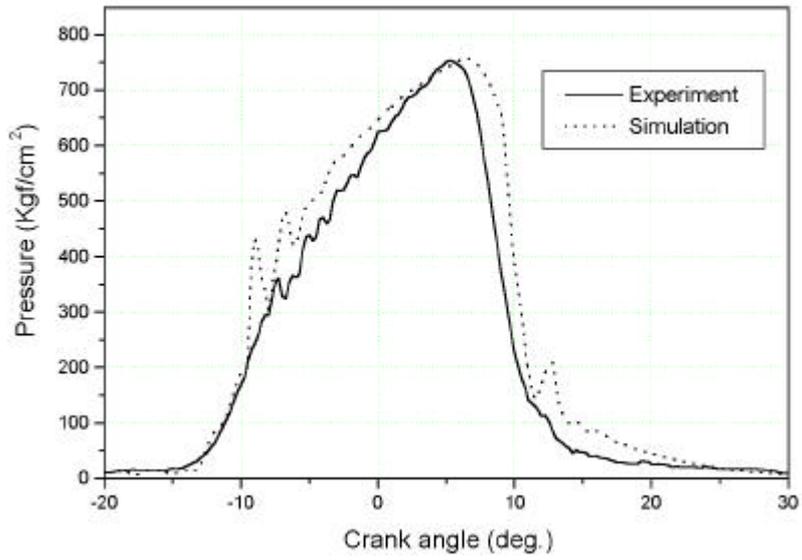


fig. 4- 4 Comparison of simulated and experimental results (RPM=190)

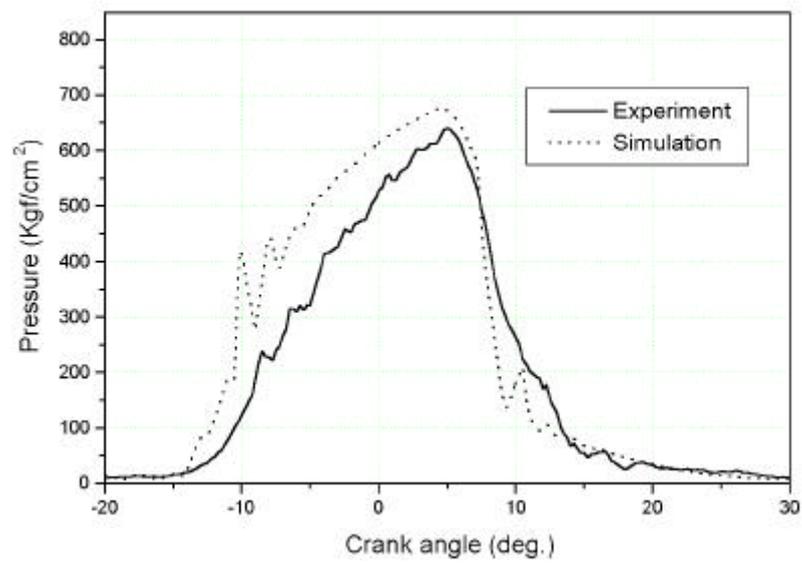


fig. 4- 5 Comparison of simulated and experimental results (RPM=180)

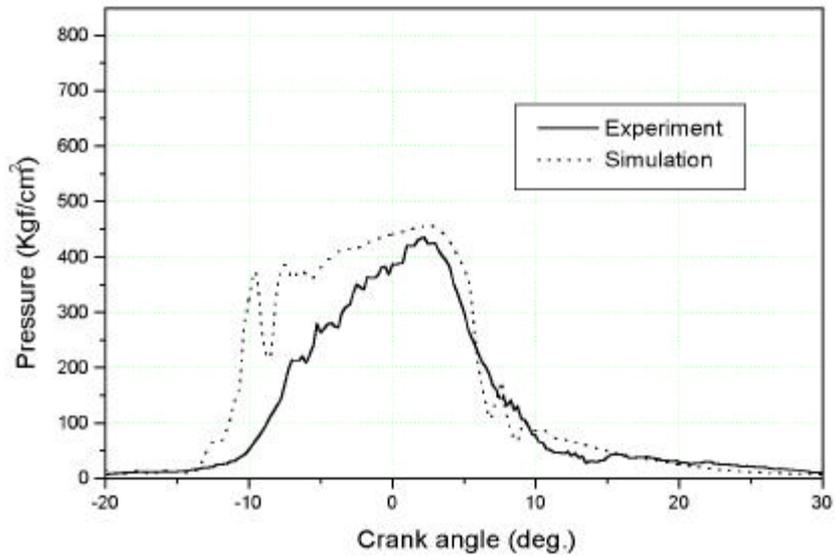


Fig. 4- 4 Comparison of simulated and experimental results (RPM=160)

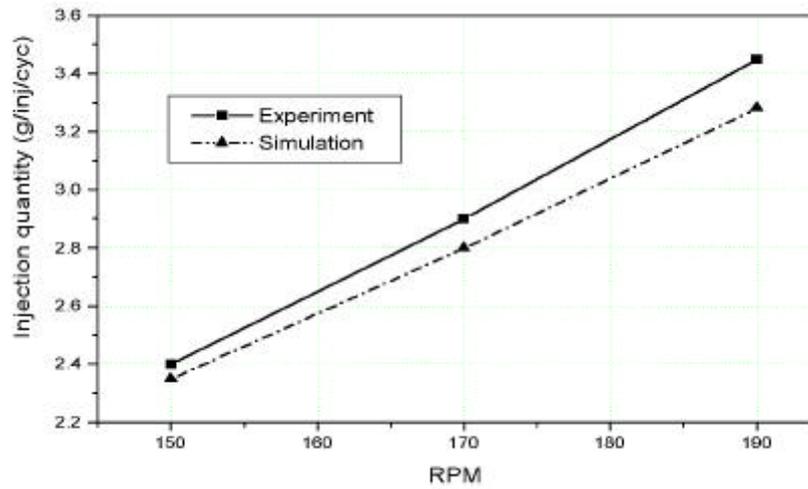


Fig. 4- 5 Comparison of simulated and experimental results

4.3

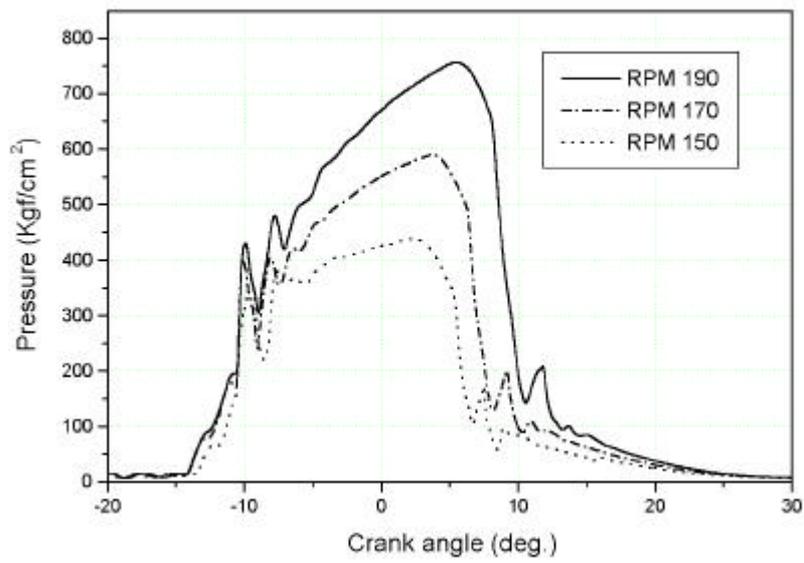
4.3.1

Fig.4- 6

RPM

Fig.4- 7

가



**fig. 4- 6 injection pressure at various engine speed
(at the distributor)**

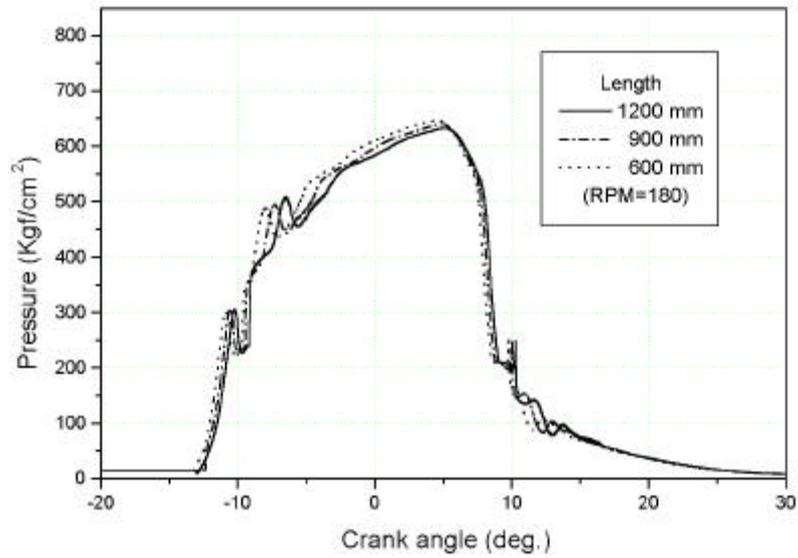


Fig. 4-6 Effect of the high pressure pipe length on injection pressure(at the nozzle chamber)

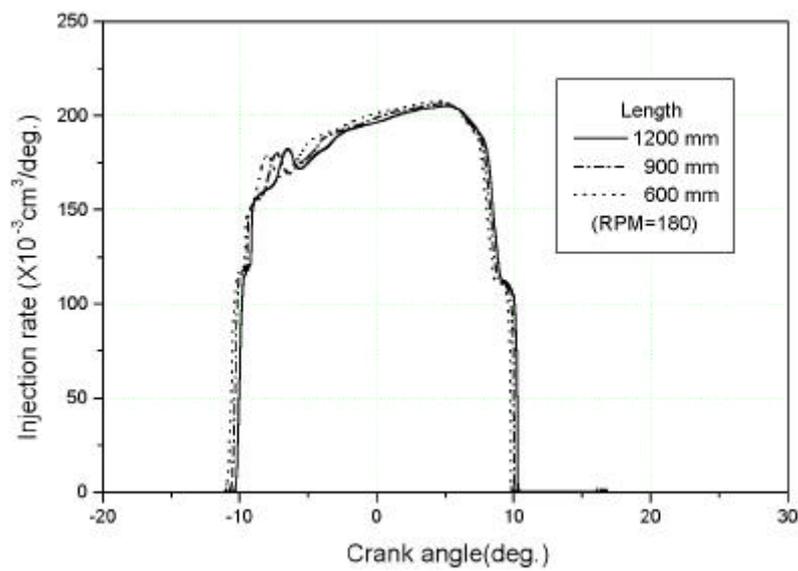


Fig. 4-7 Effect of the high pressure pipe length on injection rate

Fig.4- 10 Fig.4- 11

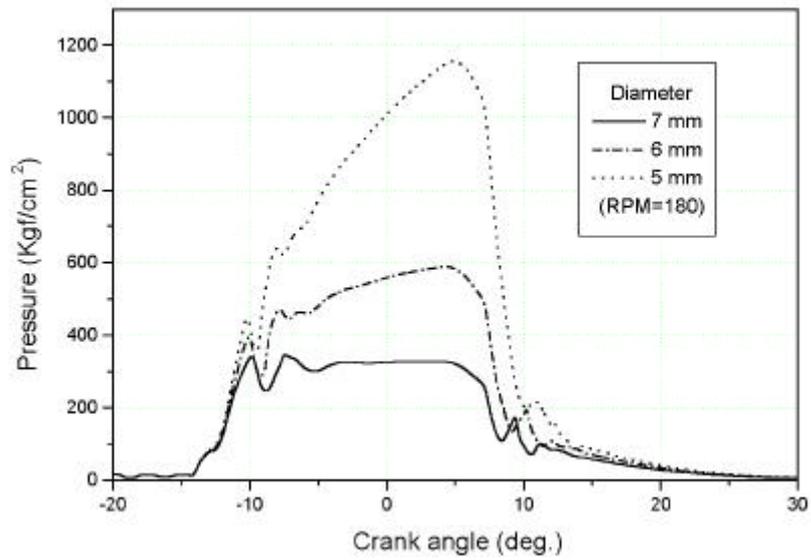


Fig. 4- 10 Effect of the high pressure pipe diameter on injection pressure (at the distributor)

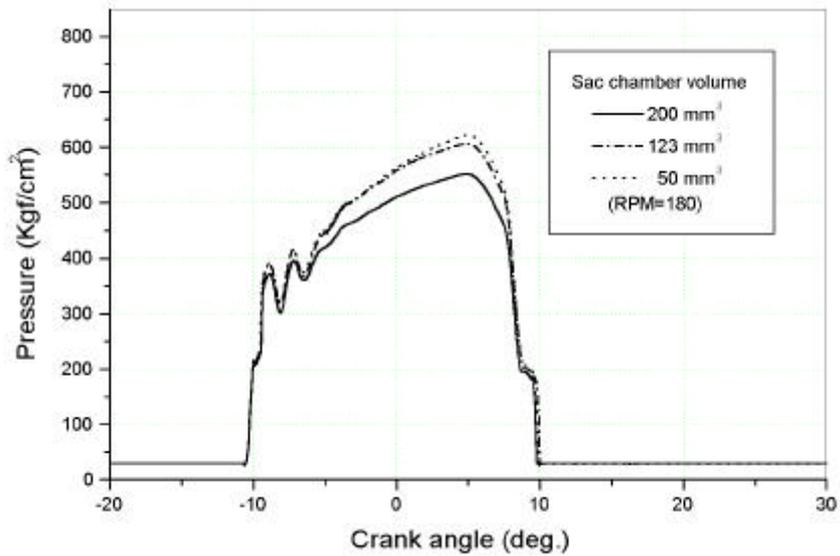


Fig. 4- 12 Effect of the sac volume on injection pressure (at the sac chamber)

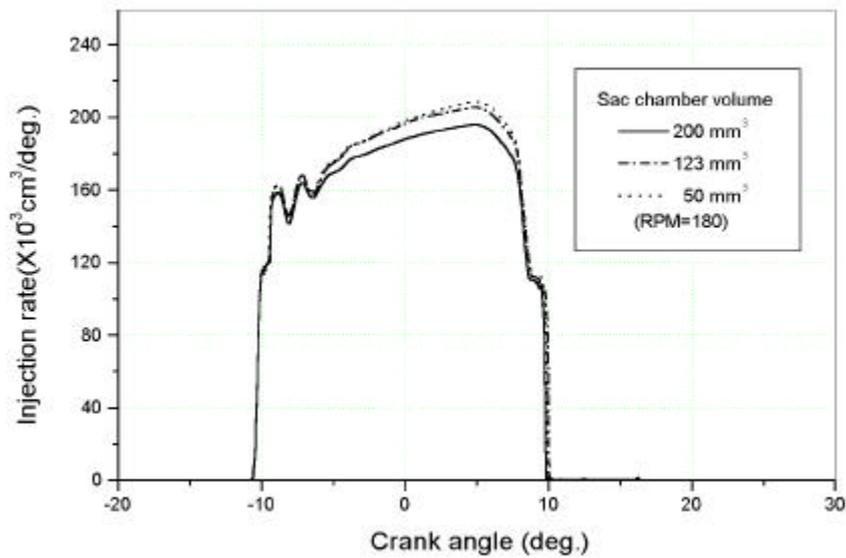


Fig. 4- 13 Effect of the sac volume on injection rate

4.3.4

Fig.4- 14, Fig.4- 15 Fig.4- 16

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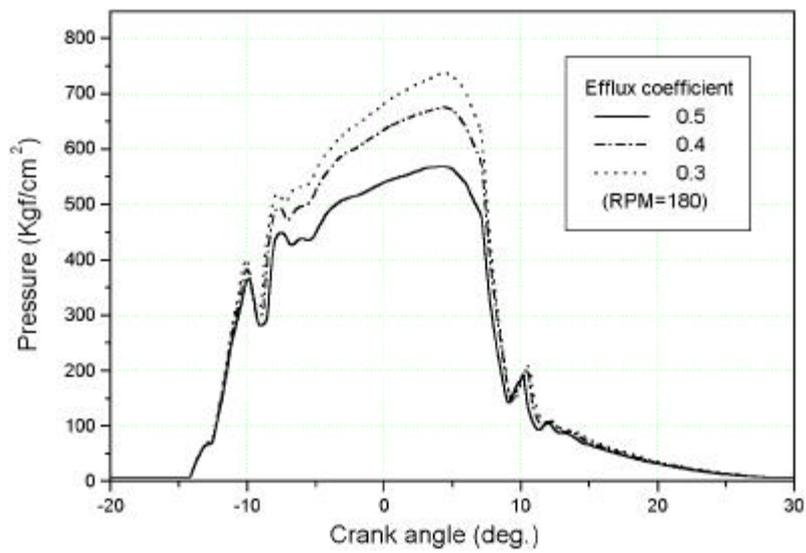


Fig. 4- 14 Effect of the efflux coefficient for pipe inlet on injection pressure

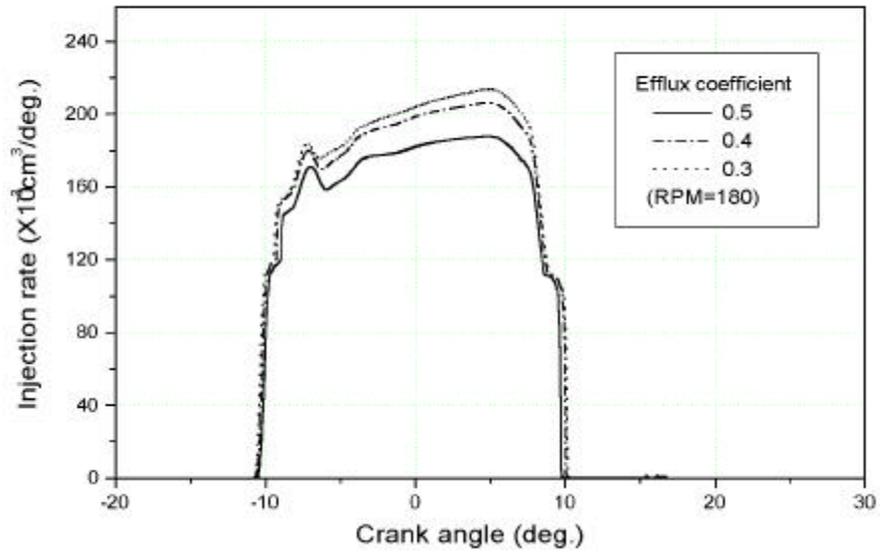


Fig. 4- 15 Effect of the efflux coefficient for pipe

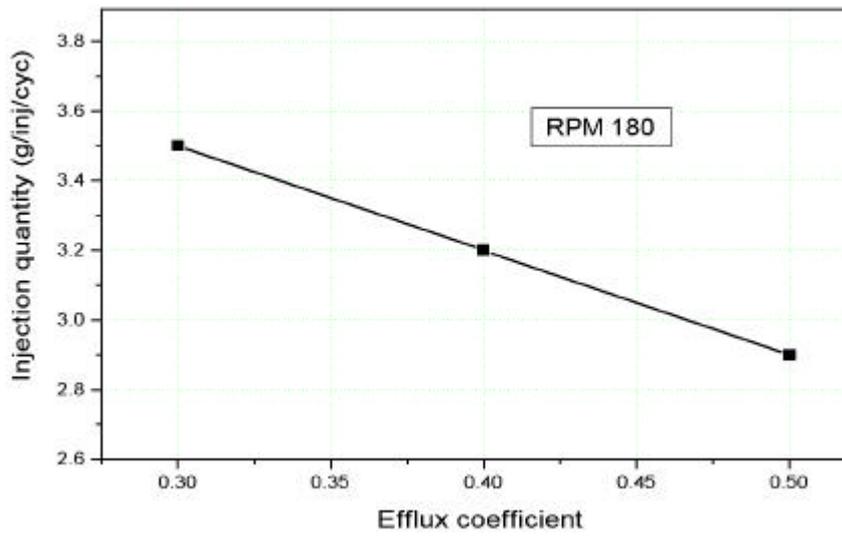


Fig. 4- 16 Effect of the efflux coefficient for pipe inlet on injection quantity

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(Sac volume)

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